



Engineering, Operations & Technology



**International Conference on Flow Physics and its Simulation
In memory of Prof. Jaw-Yen Yang**

**CFD Simulation of Serpentine S-Duct
With Flow Control**

**December 03-05, 2016
National Taiwan University**

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Outline

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Experimental & CFD Models

- ONERA test model
- Overset Grids for Half Model
- OVERFLOW RANS/URANS

Validation

- Surface Pressures
- Pressure contours at PAI

Passive Flow Control

- Effects of VG number, height, and orientation angle
- Recovery and Distortion

Active Flow Control

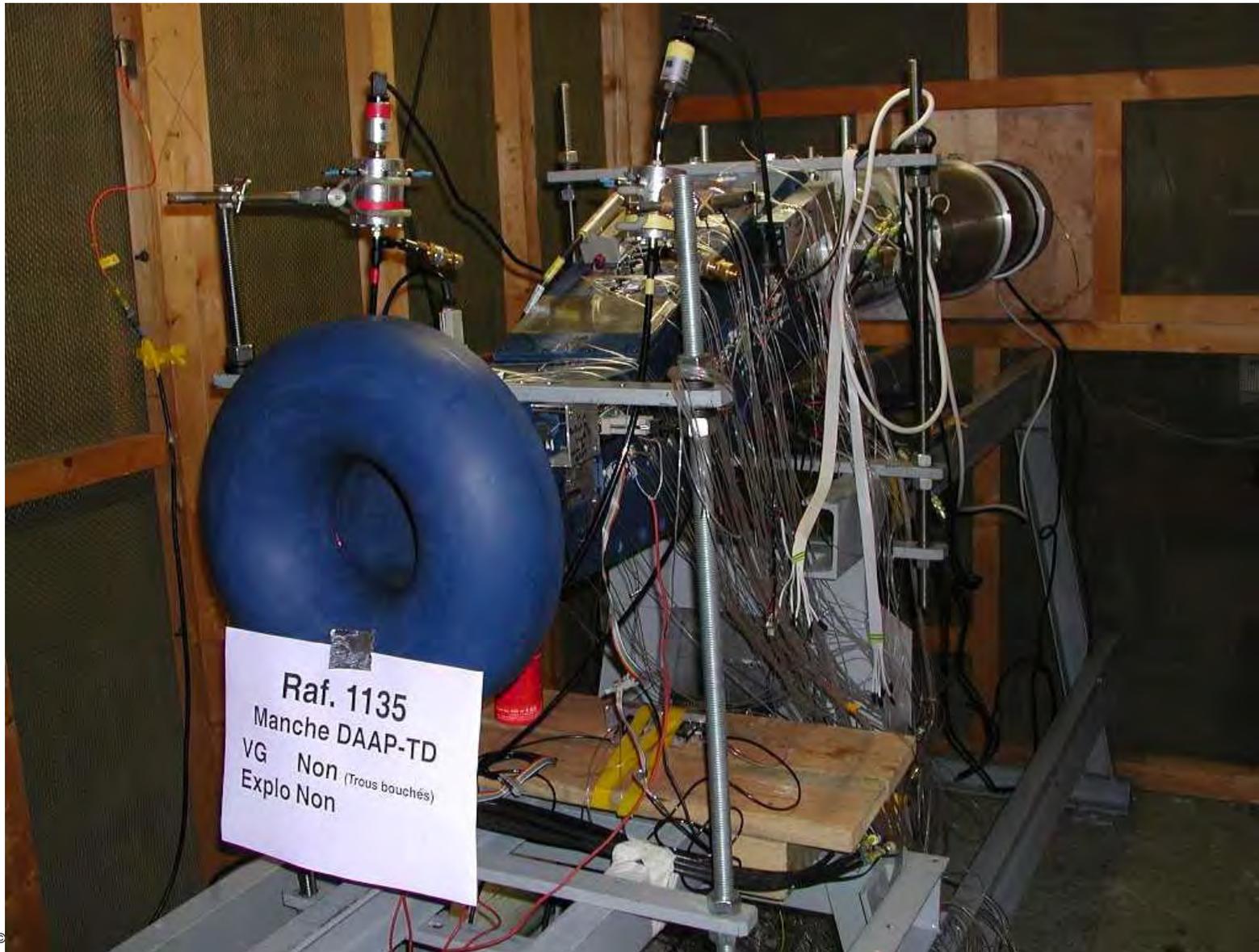
- Blowing jets model

Time Accurate Run (URANS)

Conclusions

S-duct Test Case (serpentine diffuser)

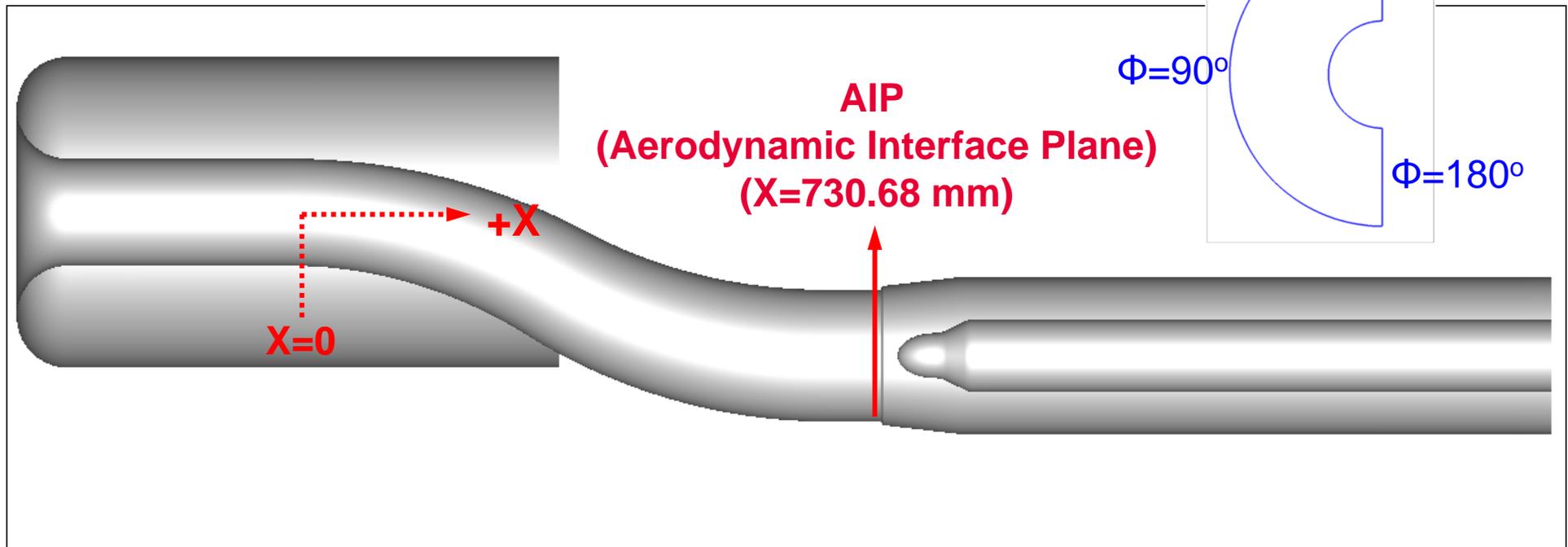
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Nomenclature

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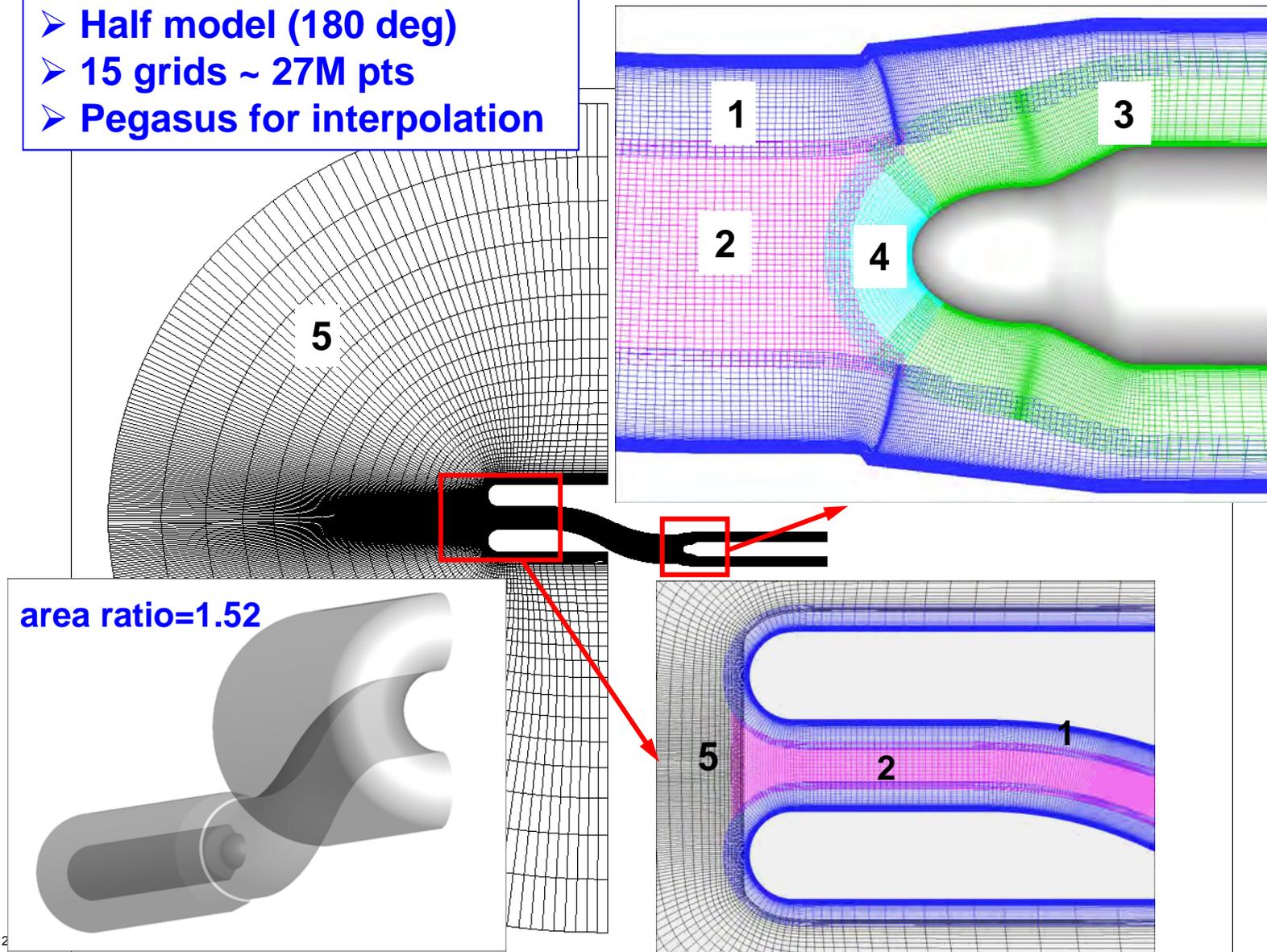
- Flow in +X direction
- Aerodynamic Interface Plane (AIP)
- Polar angle in circumferential direction (Φ)



Overset Grid System

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- Half model (180 deg)
- 15 grids ~ 27M pts
- Pegasus for interpolation

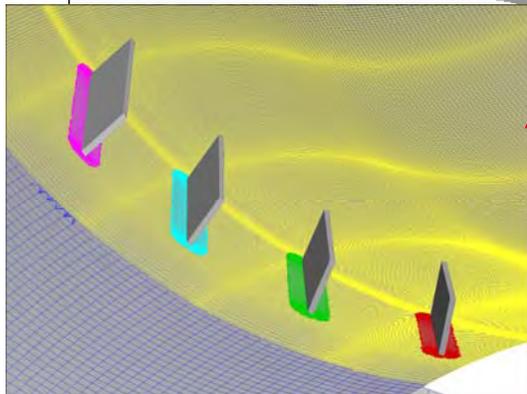


Overset Grid System

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- 4 VGs for half model
- ~1.2 M pts for each VG
- ~11 M pts for VG box

X ~ 200



- each VG consists 2 grids

18 degree relative
to free stream

6 mm height
24 mm length
0.6 mm thickness

VG cap
grid

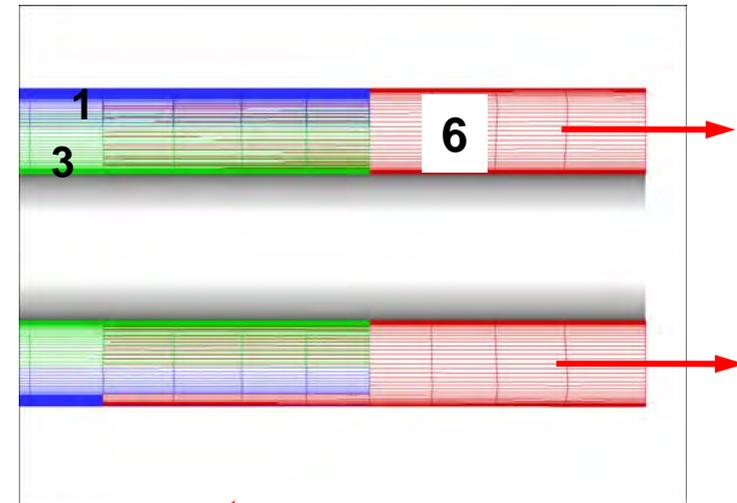
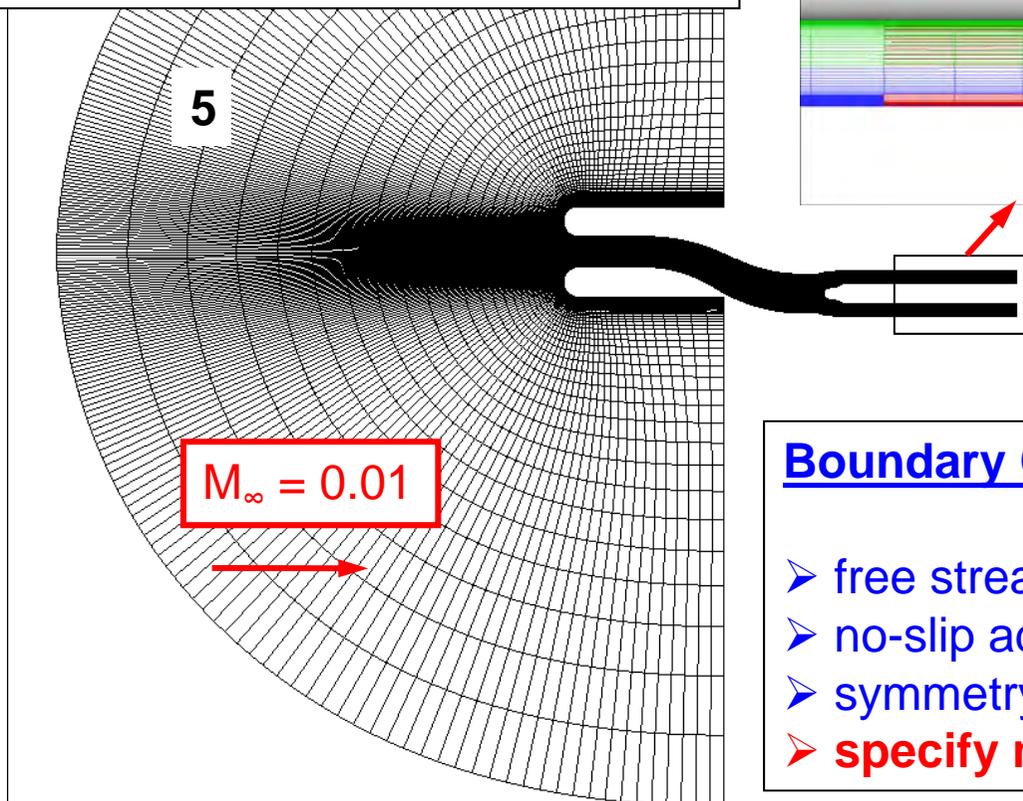
VG
grid

Flow Solver/Boundary Conditions

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OVERFLOW (RANS):

- NASA developed/overset technique
- 3 level multi-grid
- central diff, matrix dissipation
- SST turbulence model



specify mass flow:
2.427 kg/s (standard case)
1.356 kg/s (optional case)

Boundary Conditions:

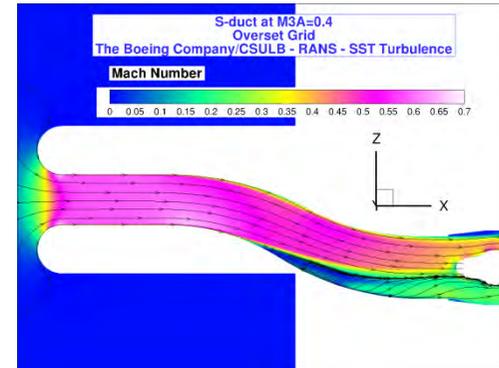
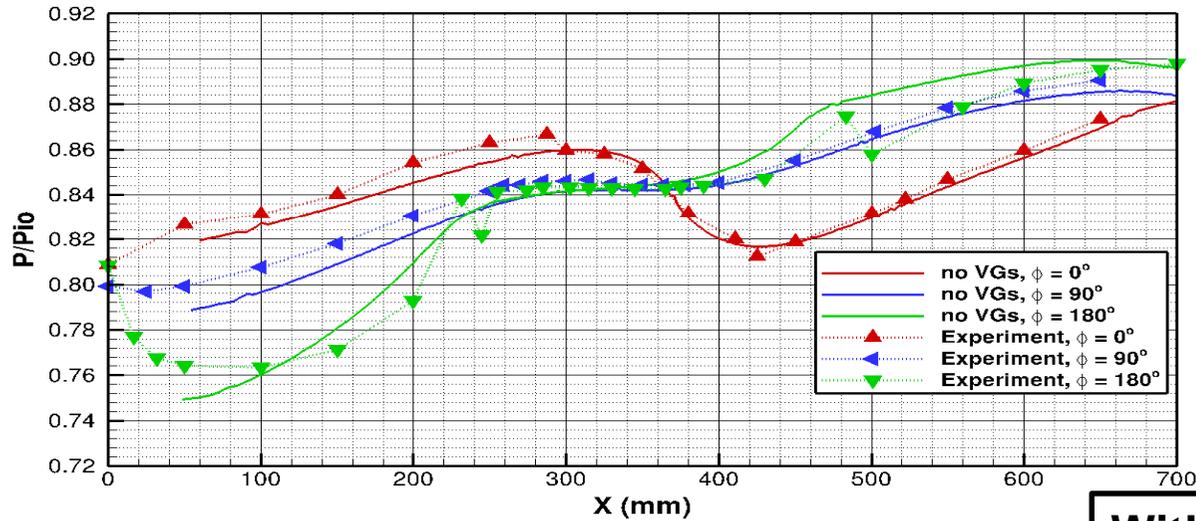
- free stream/characteristic condition
- no-slip adiabatic wall
- symmetry plane
- **specify mass flow**

Validation – Surface Pressures

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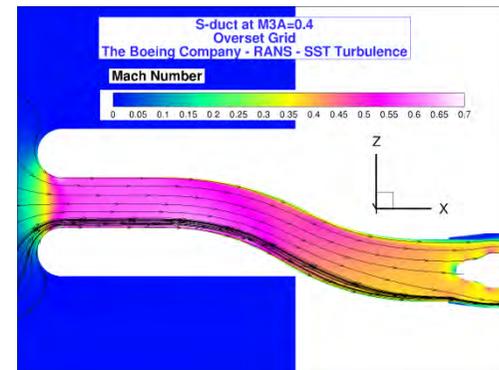
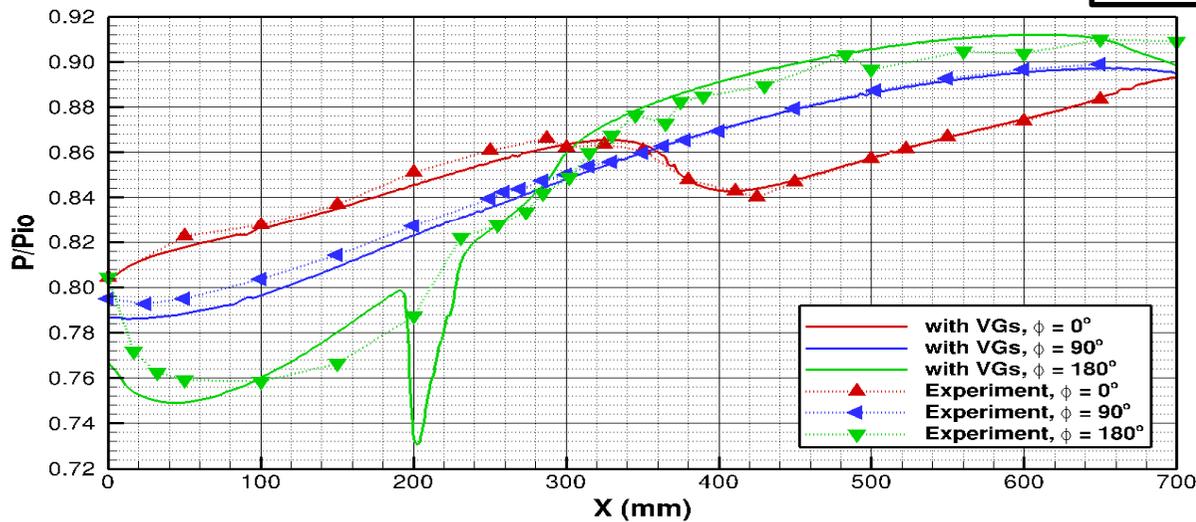
Pressure Distribution along the S-duct at $\phi = 0^\circ, 90^\circ,$ and 180°
Overset Grid (OVERFLOW)
The Boeing Company - RANS - SST Turbulence

No VGs



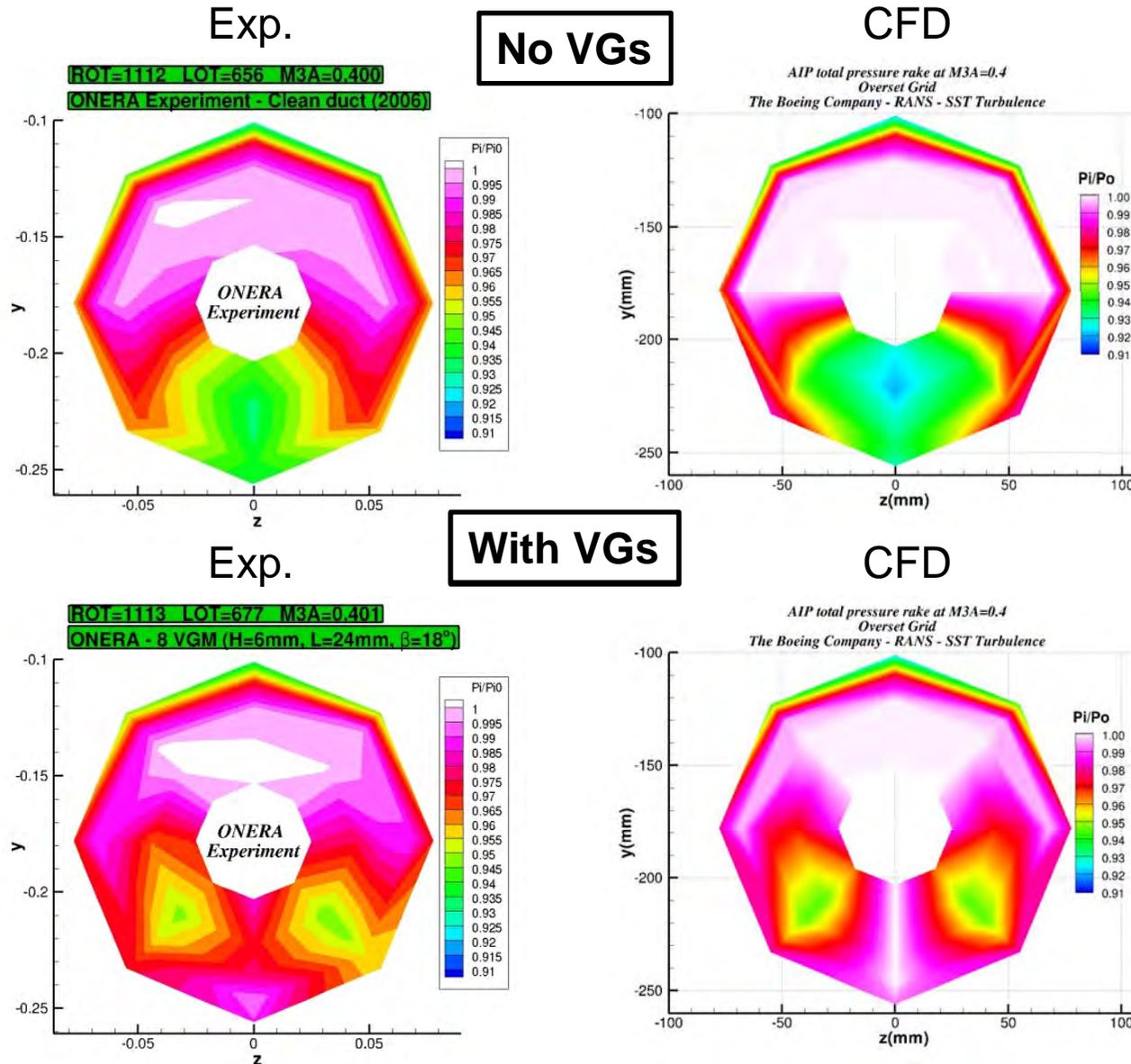
The Boeing Company - RANS - SST Turbulence

With VGs



Validation – Pressure Contours at AIP

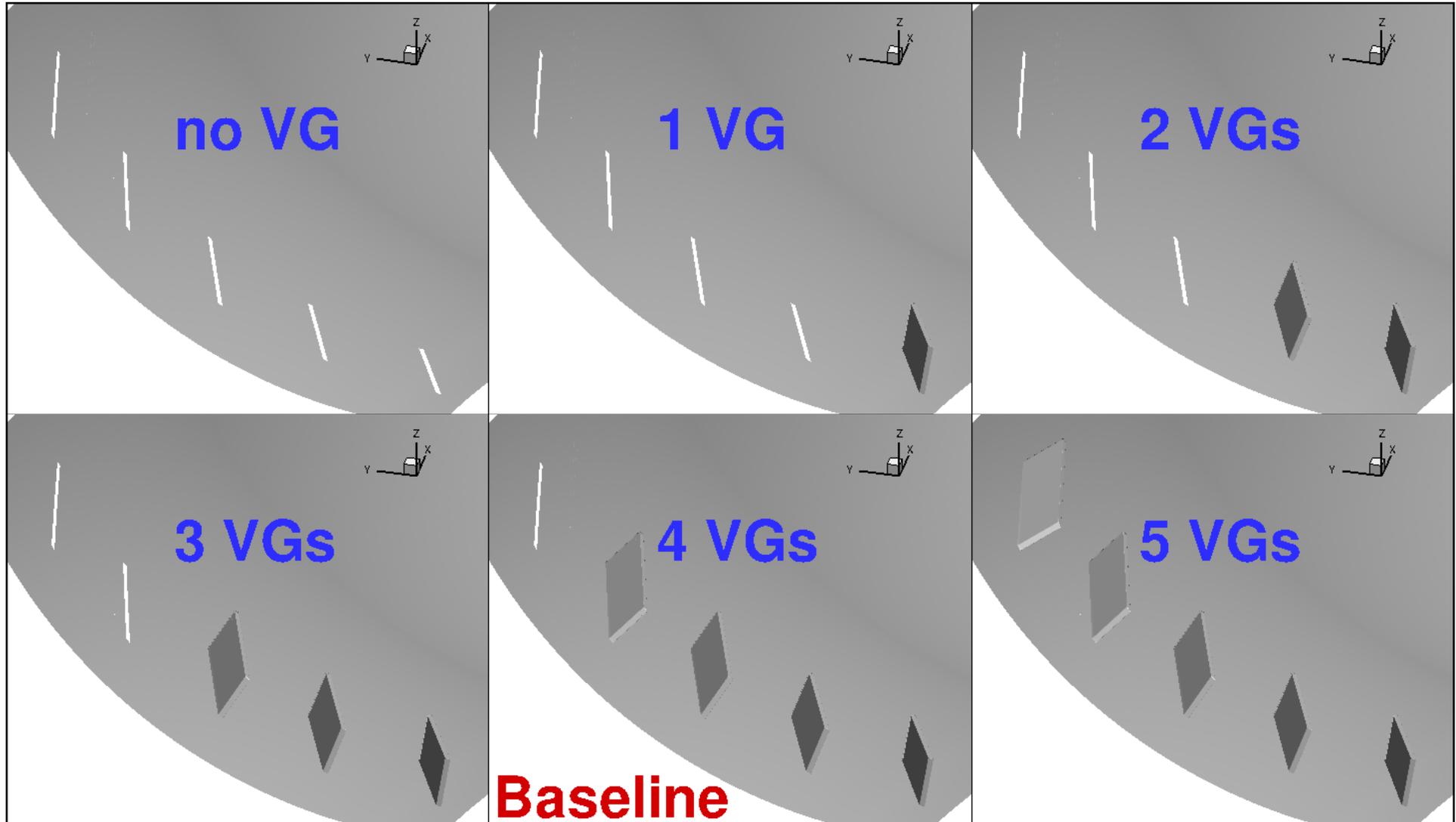
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Passive Flow Control - VGs

Passive Flow Control - VG Number Effect

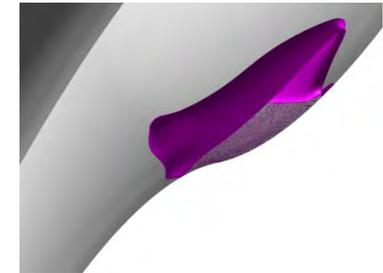
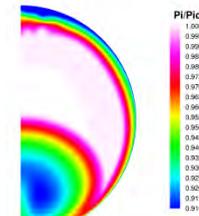
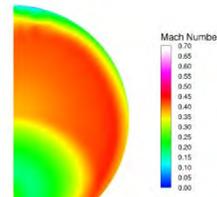
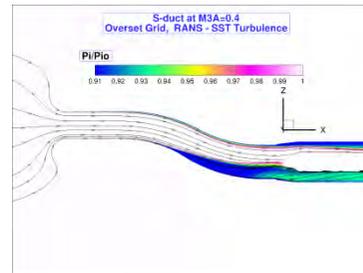
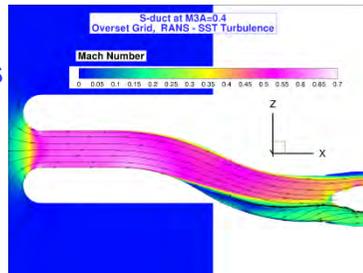
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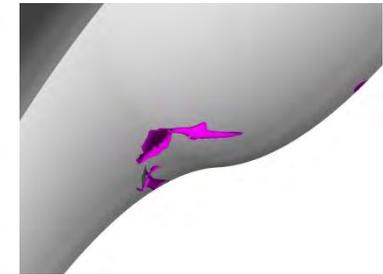
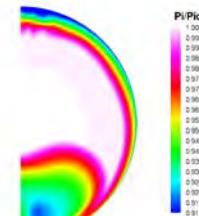
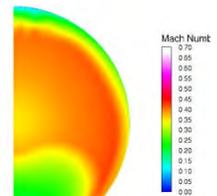
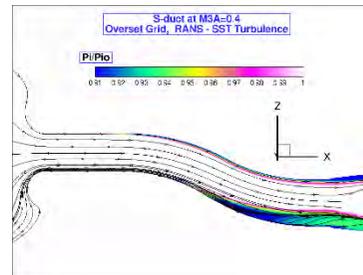
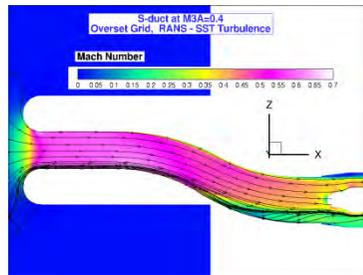
Passive Flow Control - VG Number Effect

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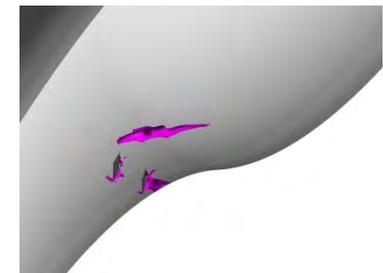
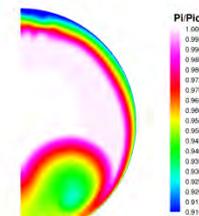
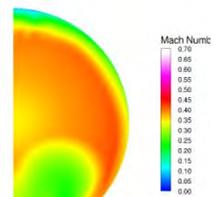
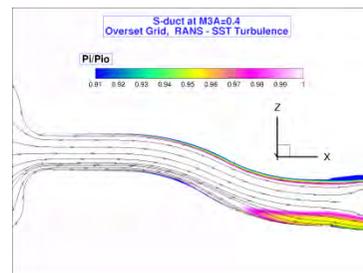
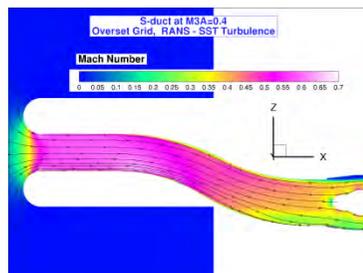
No VGs



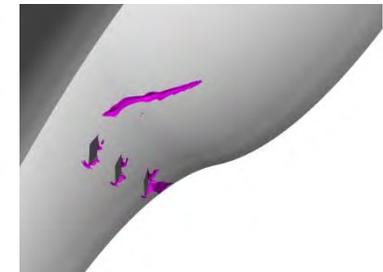
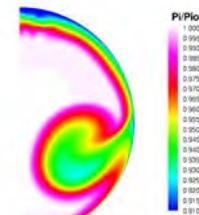
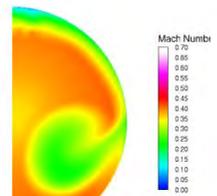
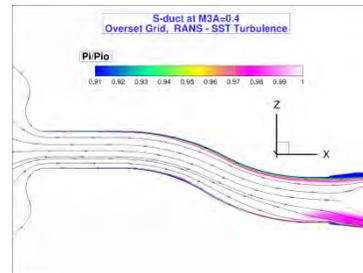
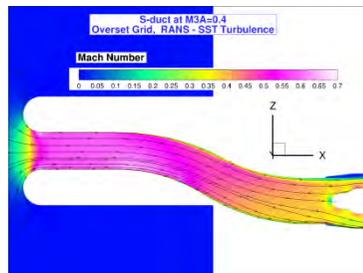
1 VGs



2 VGs



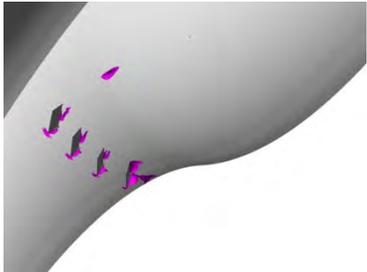
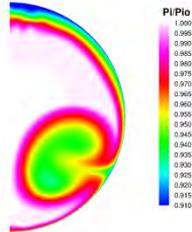
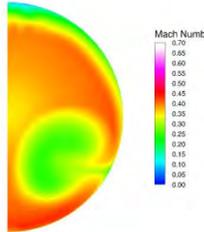
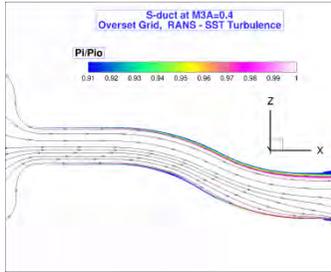
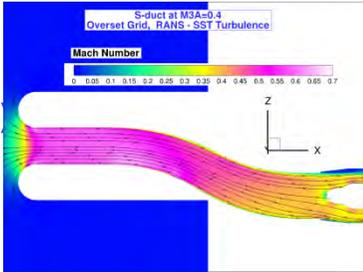
3 VGs



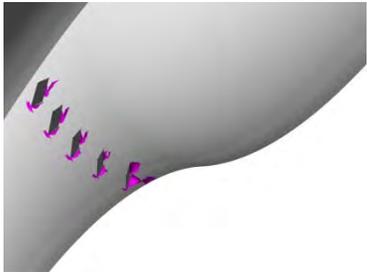
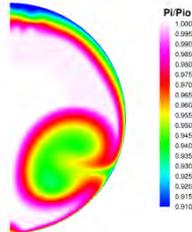
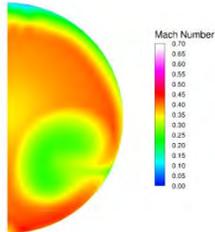
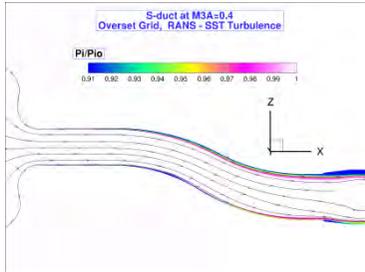
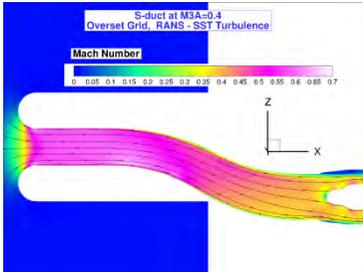
Passive Flow Control - VG Number Effect

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4 VGs
baseline



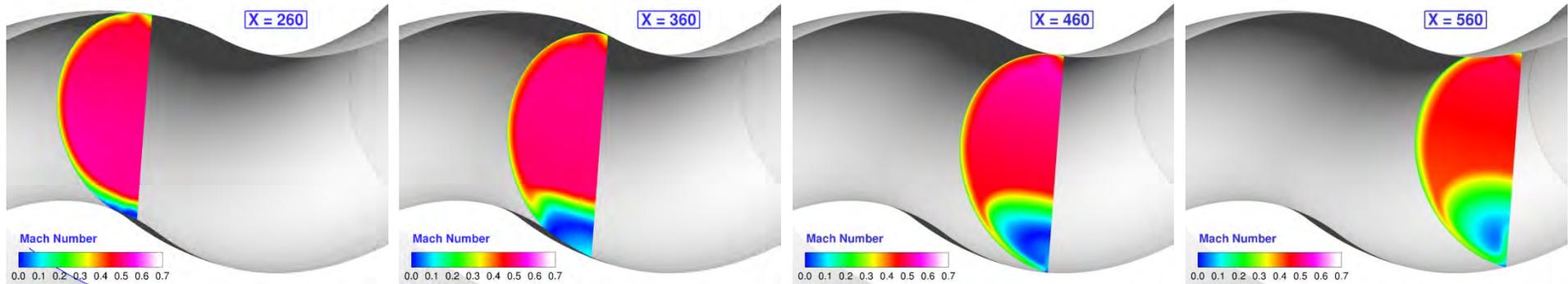
5VGs



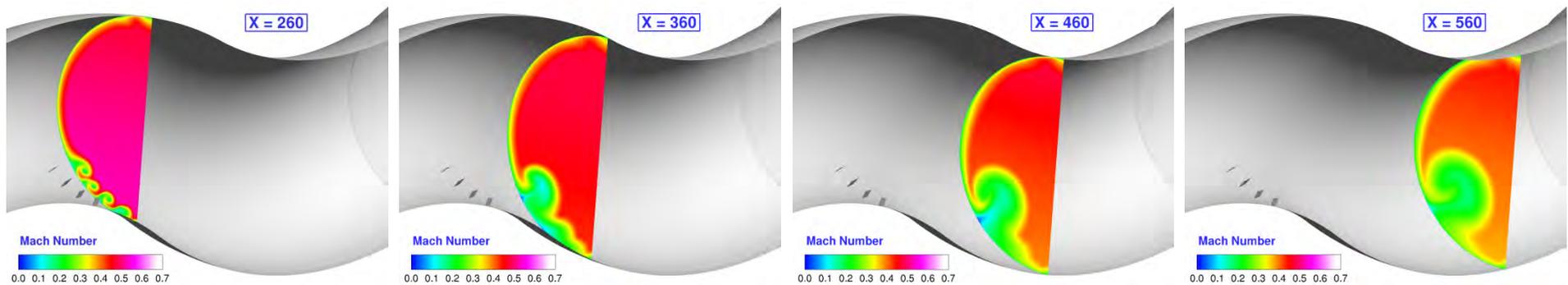
Mach contours at different X cut No VGs vs With VGs (mass flow = 2.427 kg/s)

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No VGs

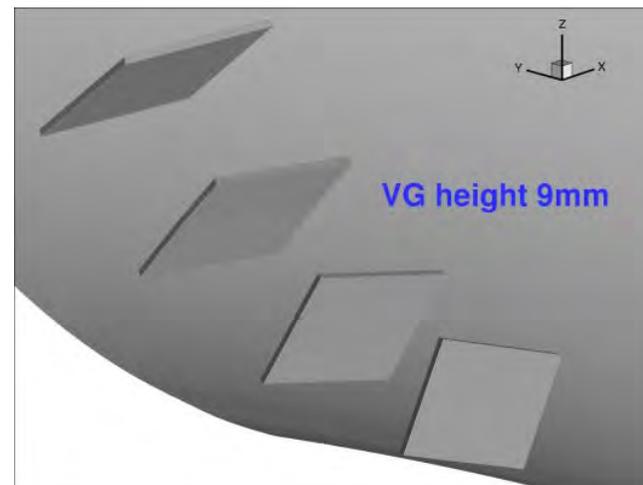
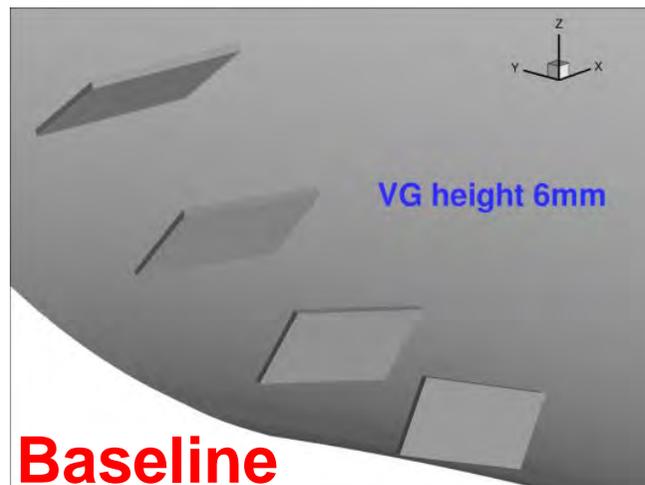
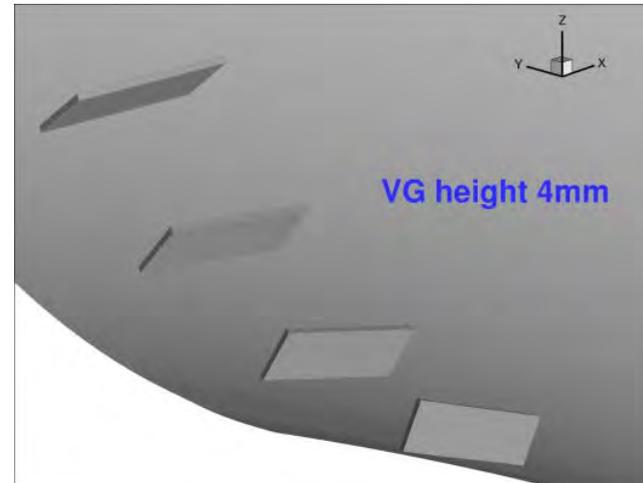
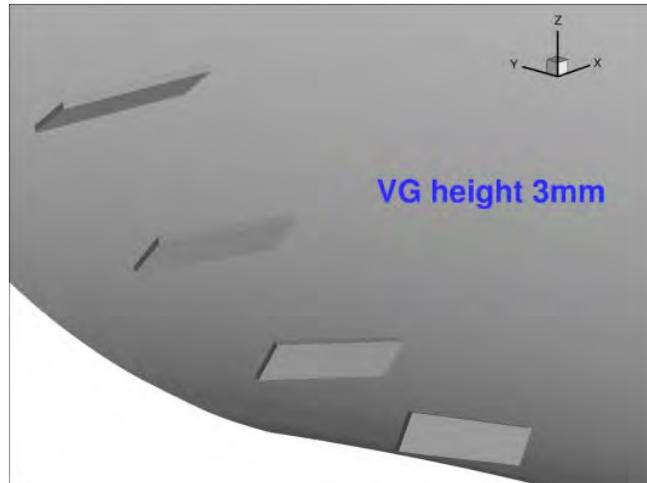


With VGs



Passive Flow Control - VG Height Effect

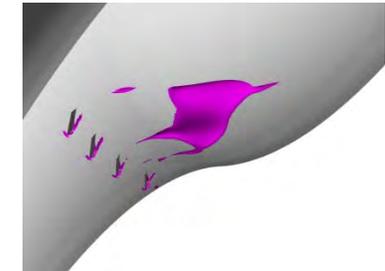
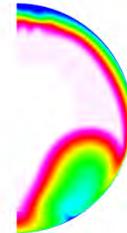
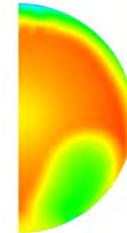
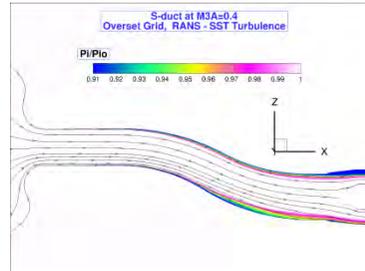
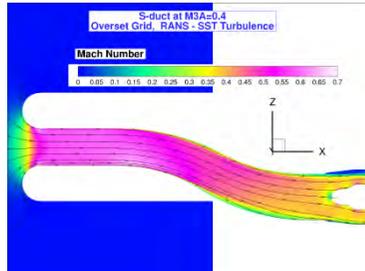
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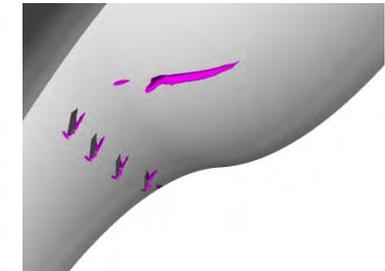
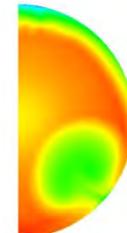
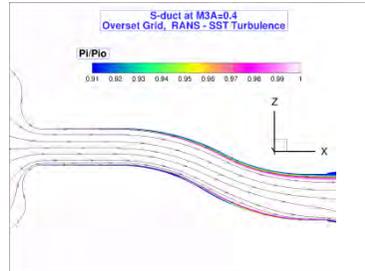
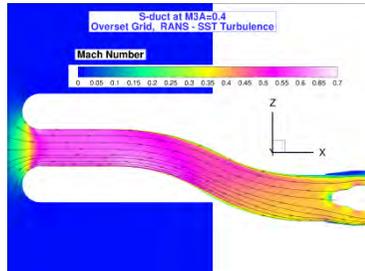
Passive Flow Control - VG height Effect

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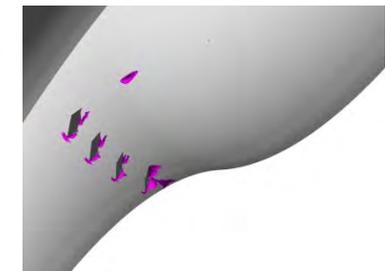
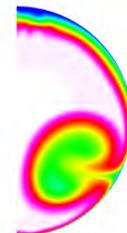
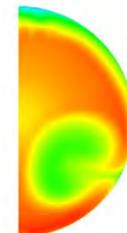
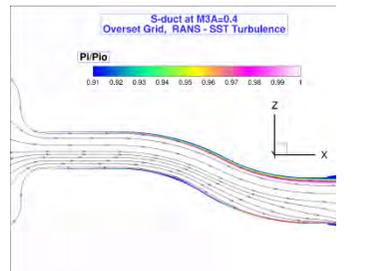
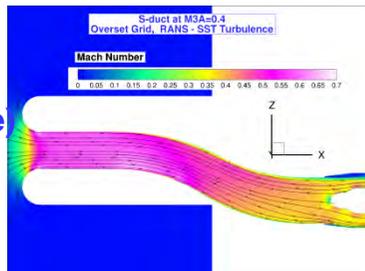
3mm



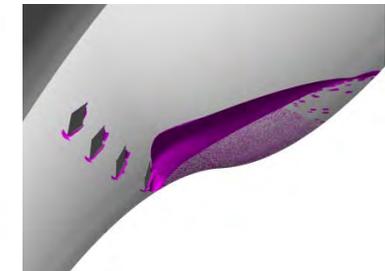
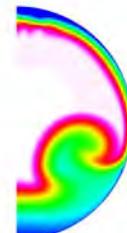
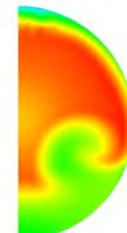
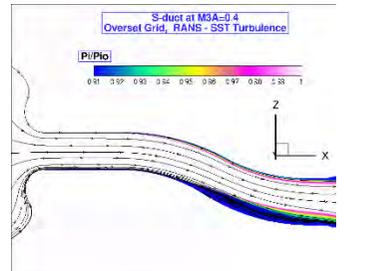
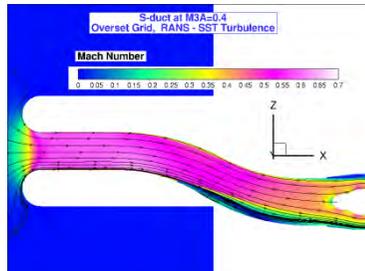
4mm



6mm
baseline

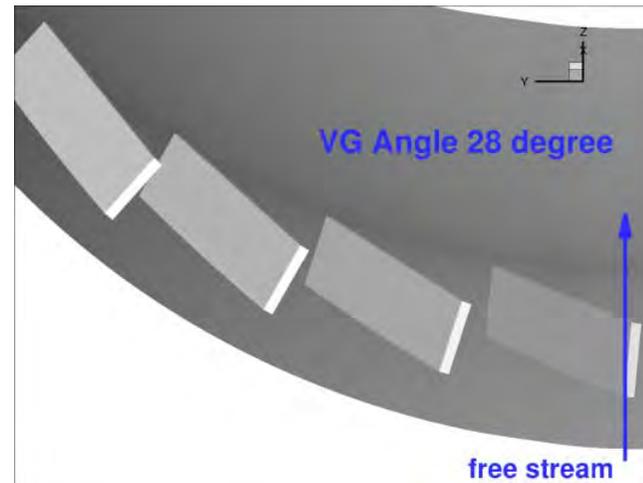
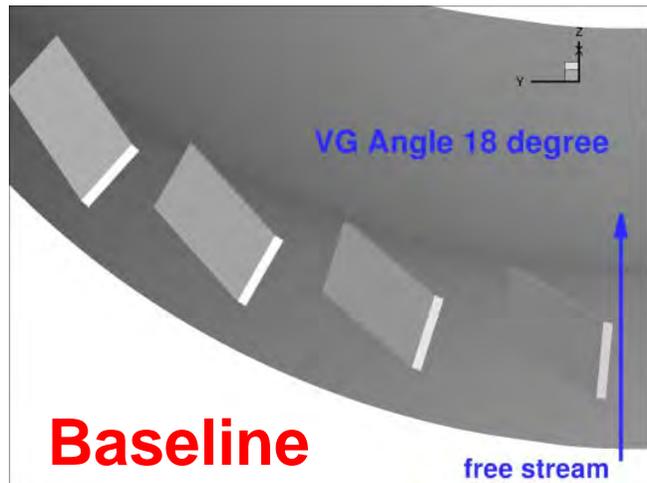
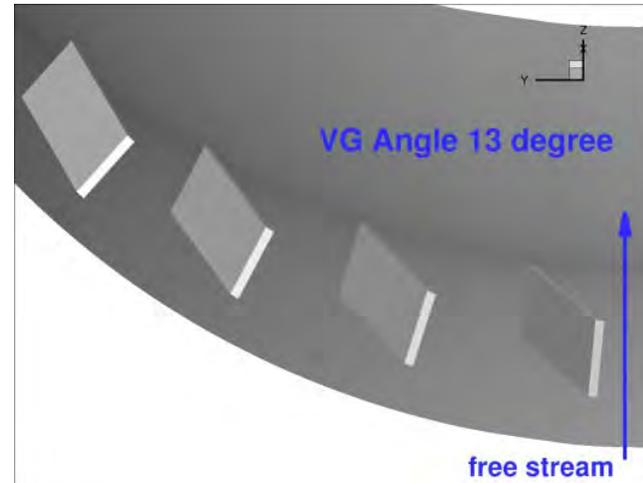
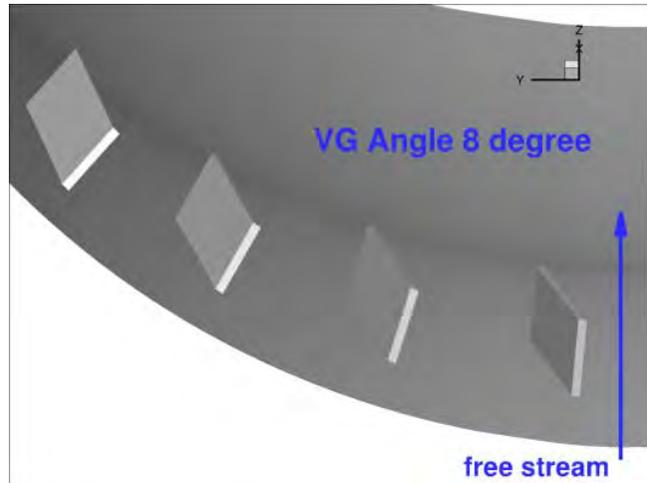


9mm



Passive Flow Control – VG Angle Effect

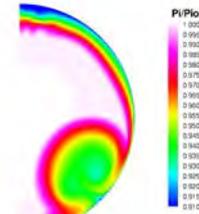
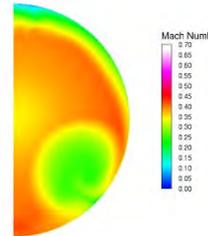
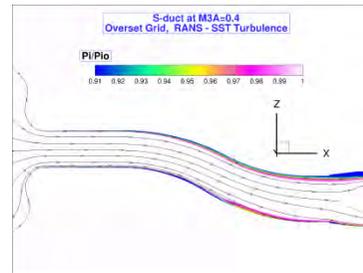
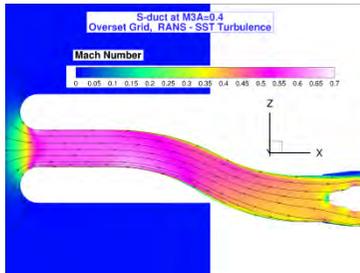
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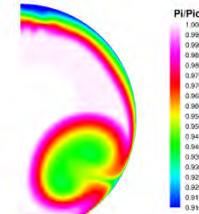
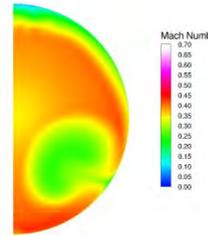
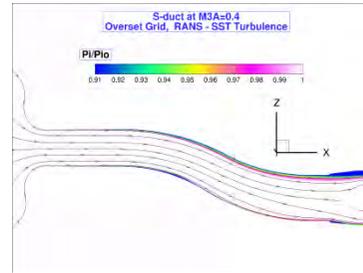
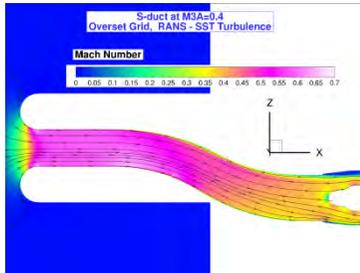
Passive Flow Control - VG Angle Effect

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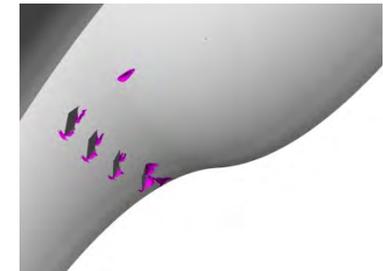
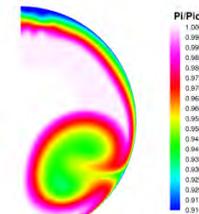
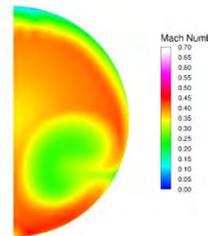
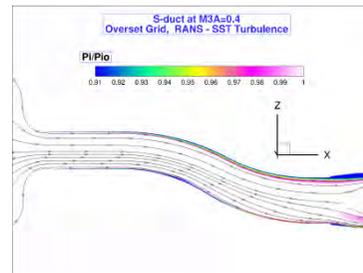
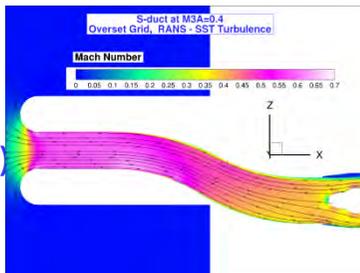
VG@
8 deg



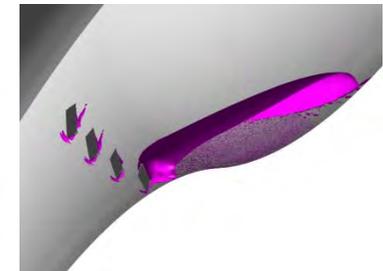
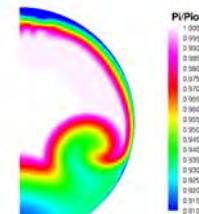
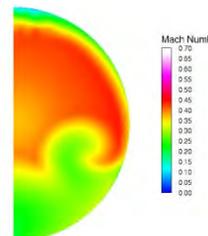
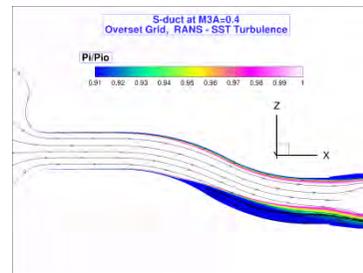
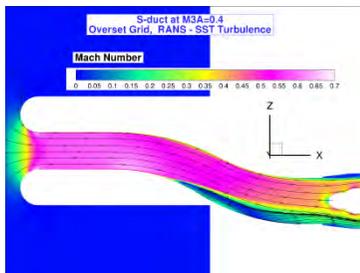
VG@
13 deg



VG@
18 deg
baseline

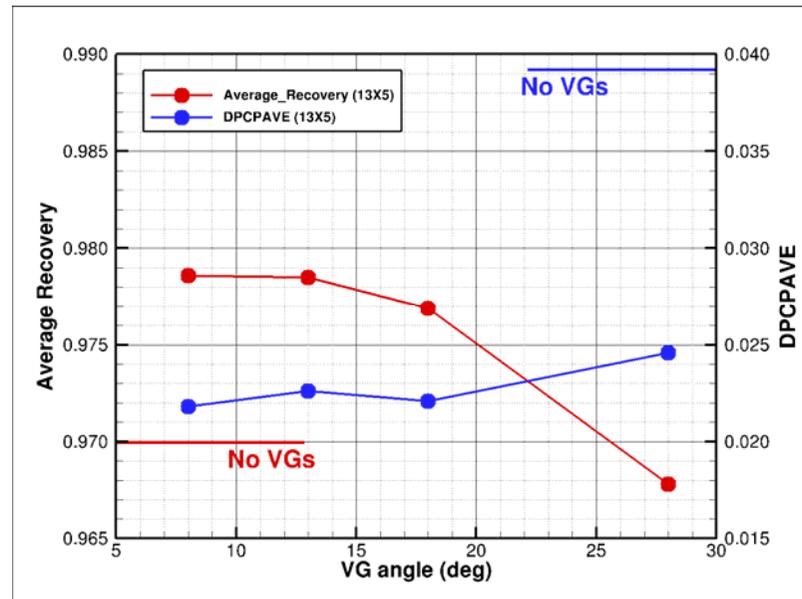
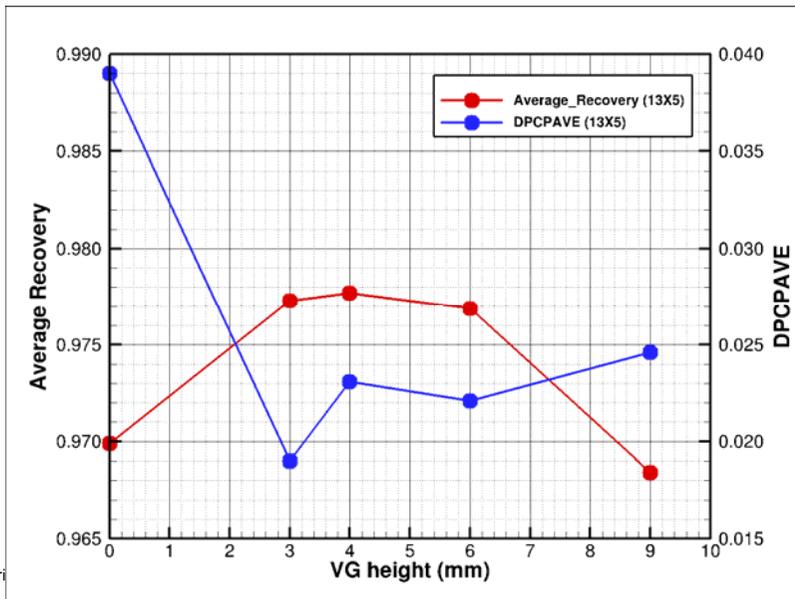
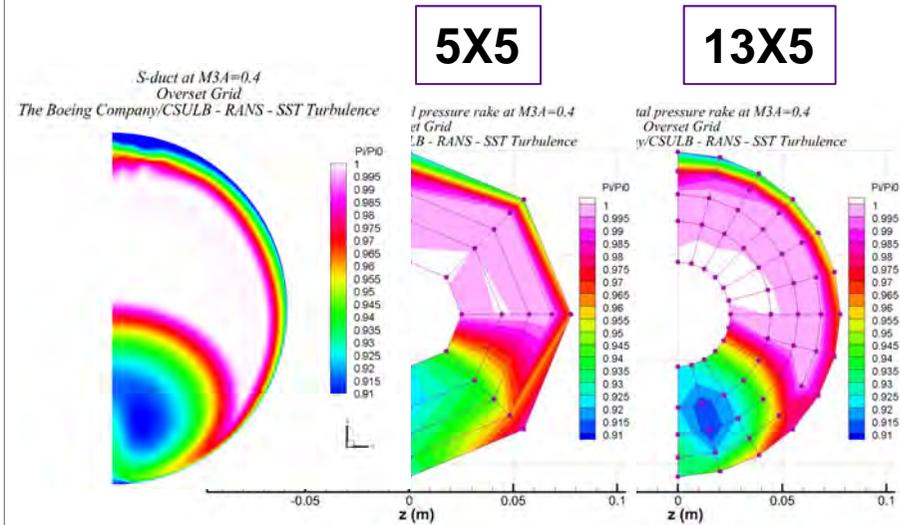
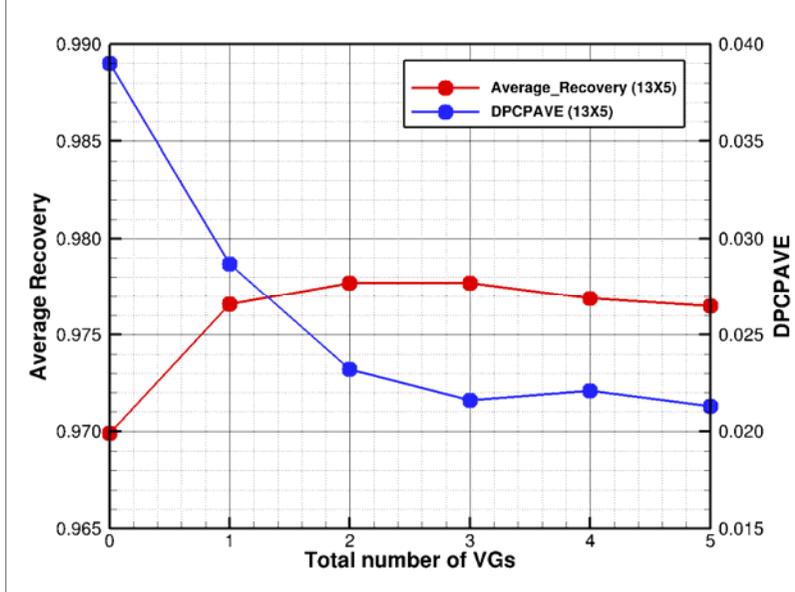


VG@
28 deg



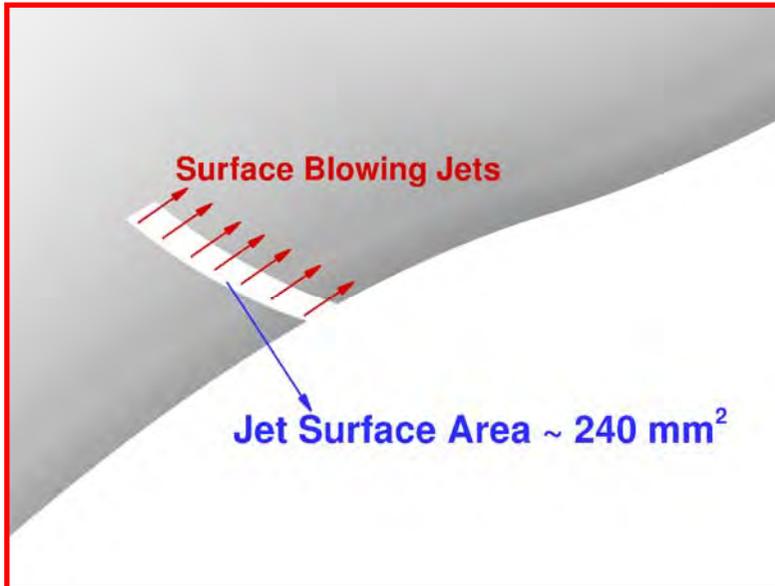
Passive Flow Control - VGs

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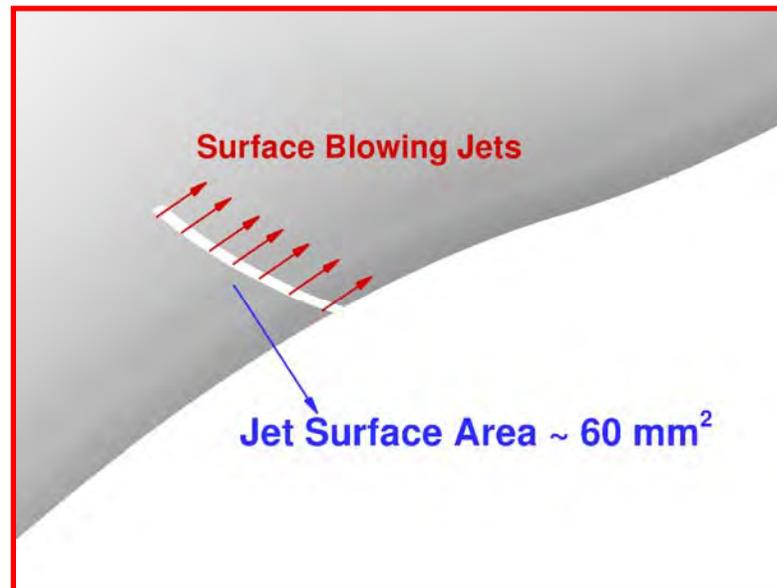
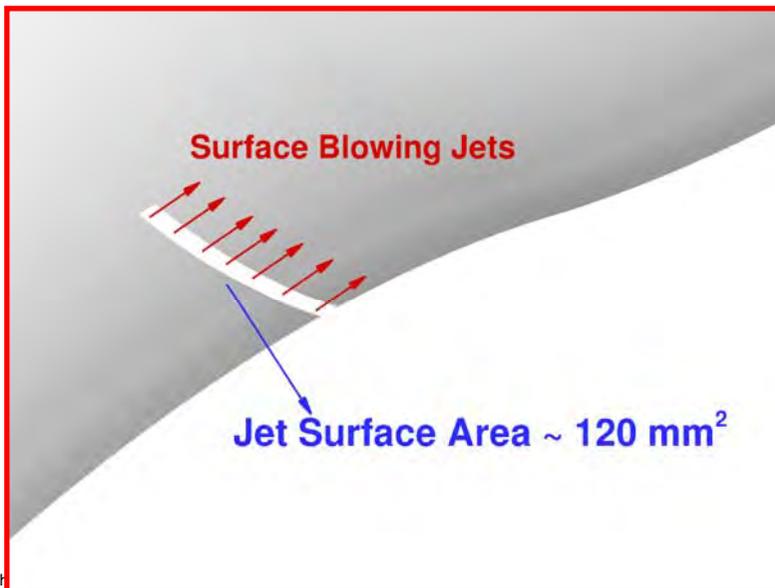


Active Flow Control – Surface Blowing Jets

Engineering, Operations & Technology



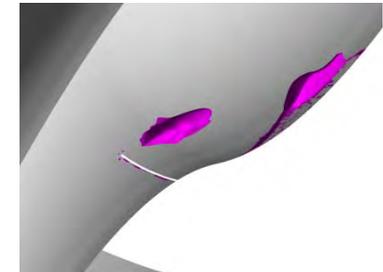
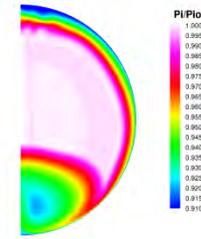
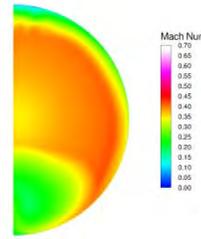
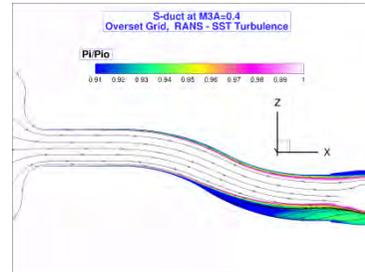
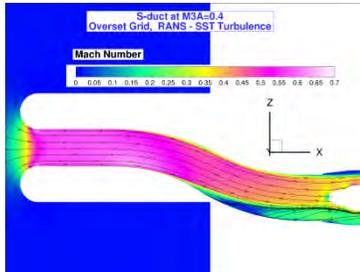
- Blowing jets applied at surface with pre-defined velocity and direction
- Jet velocity is controlled by P/P_{∞}
- Mass flow rate of the jets is decided by the jet velocity and area it applies
- **Three AFC configurations (areas) each with three different mass flow rate are simulated**



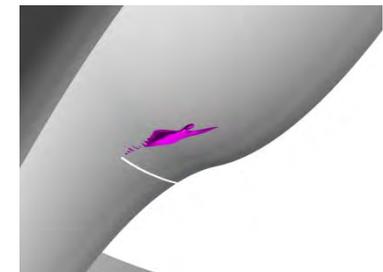
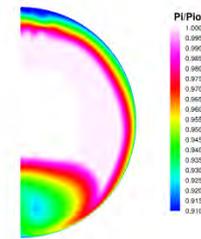
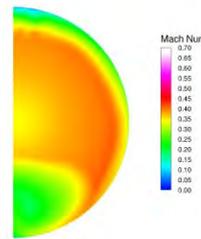
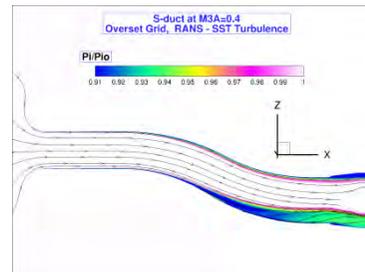
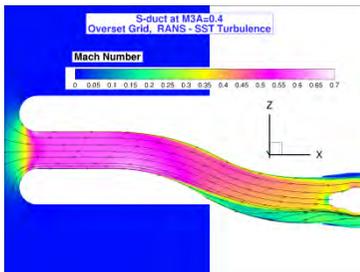
Active Flow Control – Surface Blowing Jets

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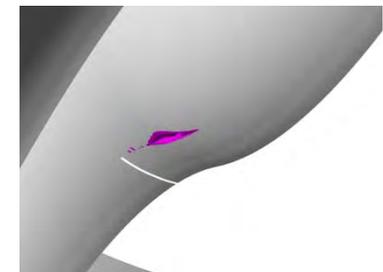
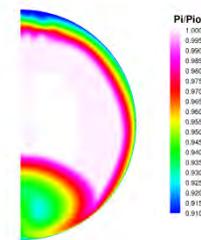
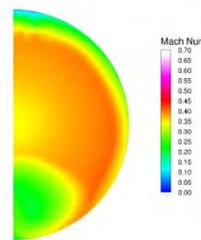
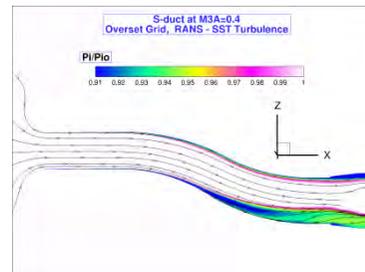
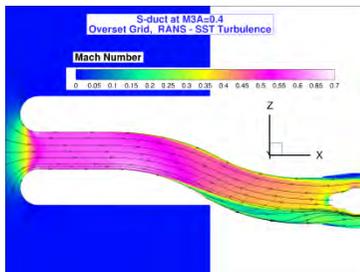
60mm²
Pt22.4



60mm²
Pt29.2



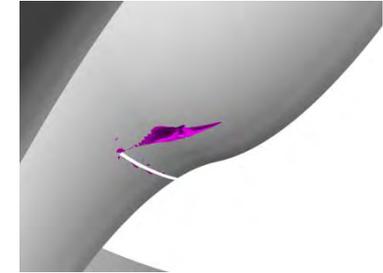
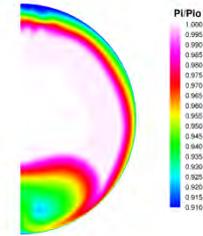
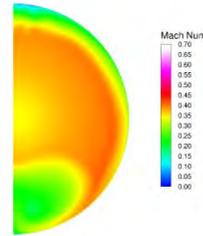
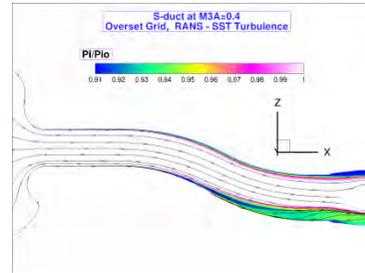
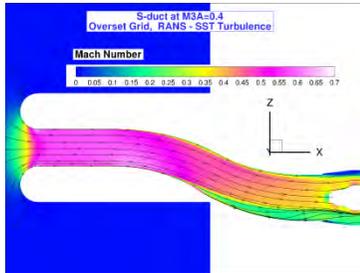
60mm²
Pt36.5



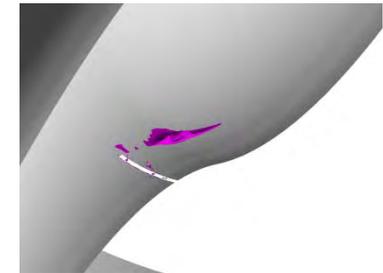
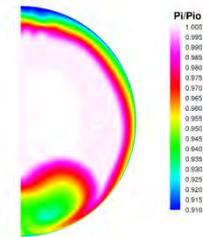
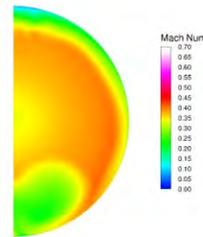
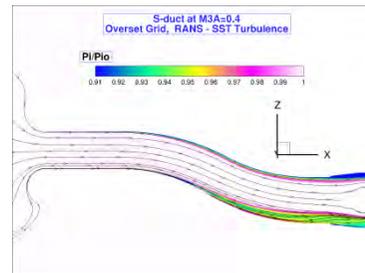
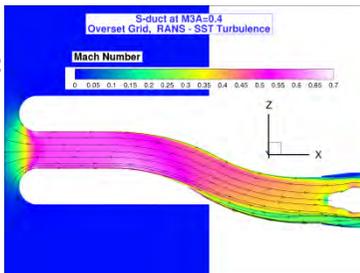
Active Flow Control – Surface Blowing Jets

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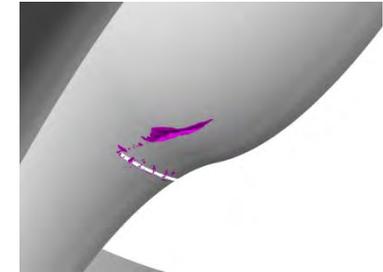
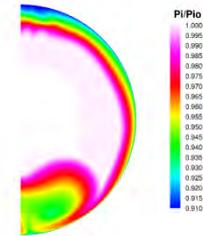
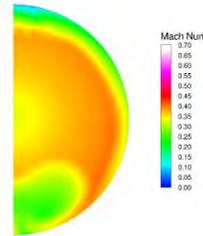
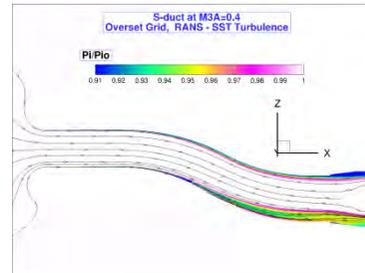
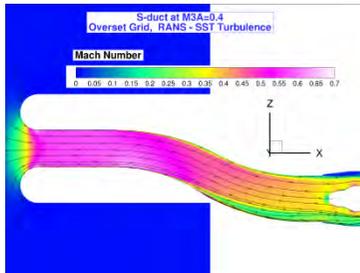
120mm²
Pt22.4



120mm²
Pt29.2



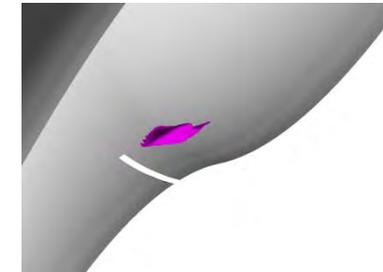
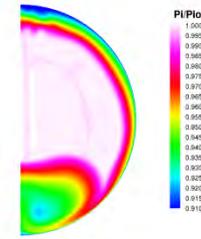
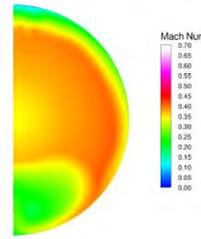
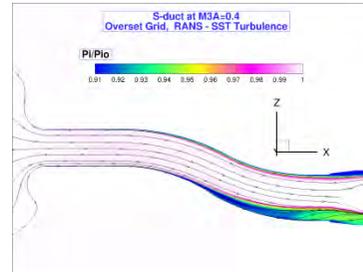
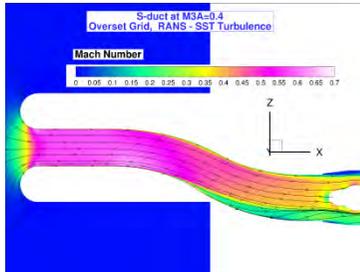
120mm²
Pt36.5



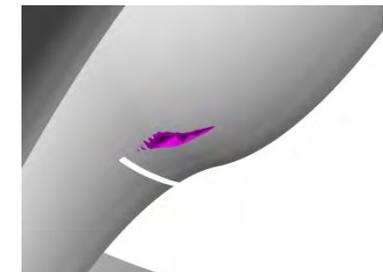
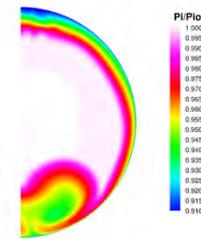
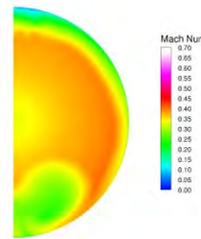
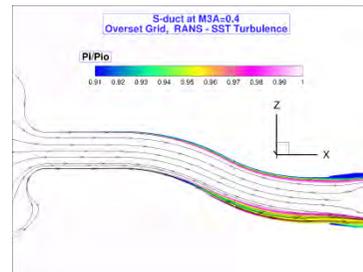
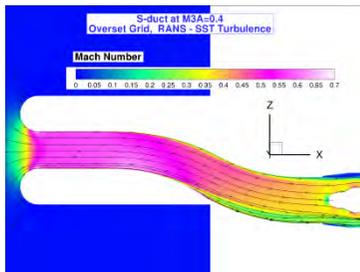
Active Flow Control – Surface Blowing Jets

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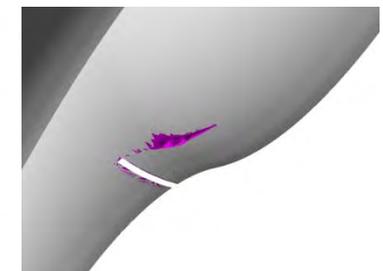
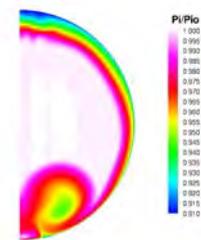
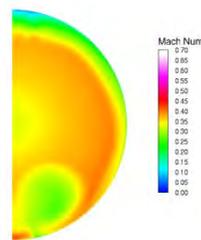
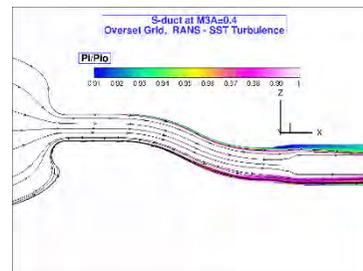
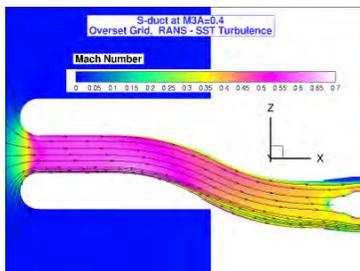
240mm²
Pt14.6



240mm²
Pt22.4

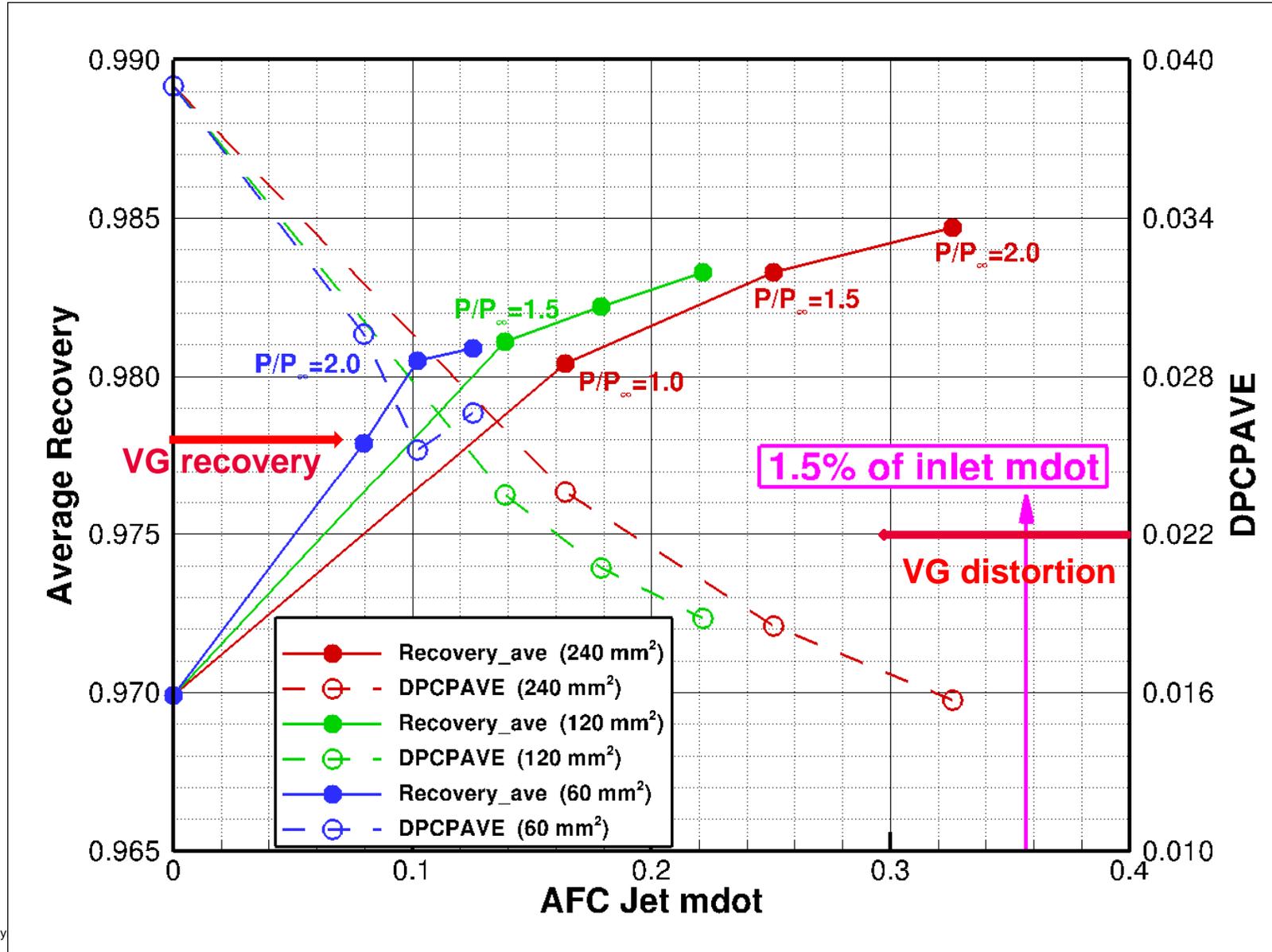


240mm²
Pt29.2



Active Flow Control – Surface Blowing Jets

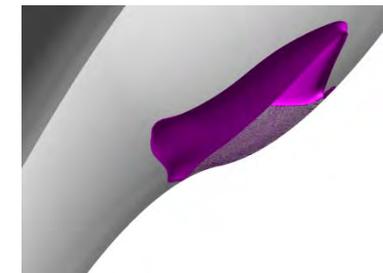
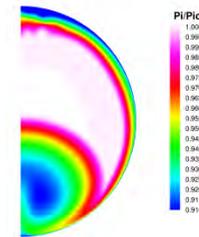
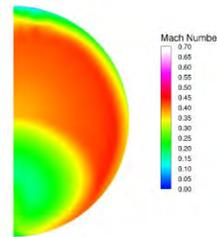
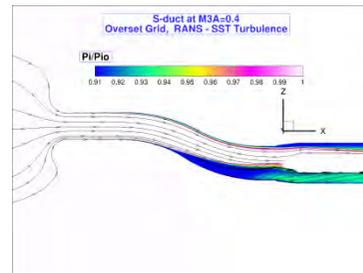
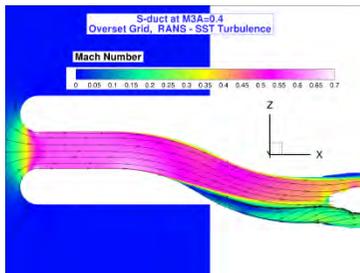
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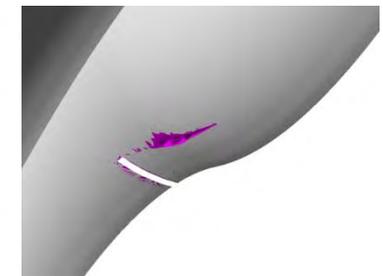
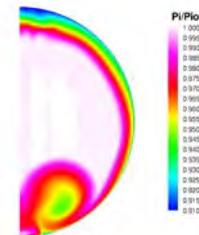
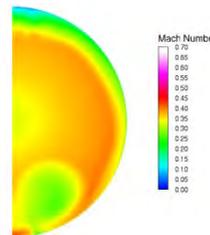
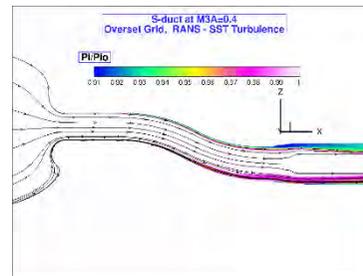
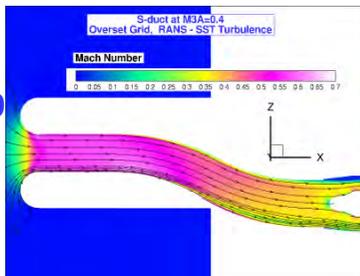
Active Flow Control – Surface Blowing Jets

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baseline



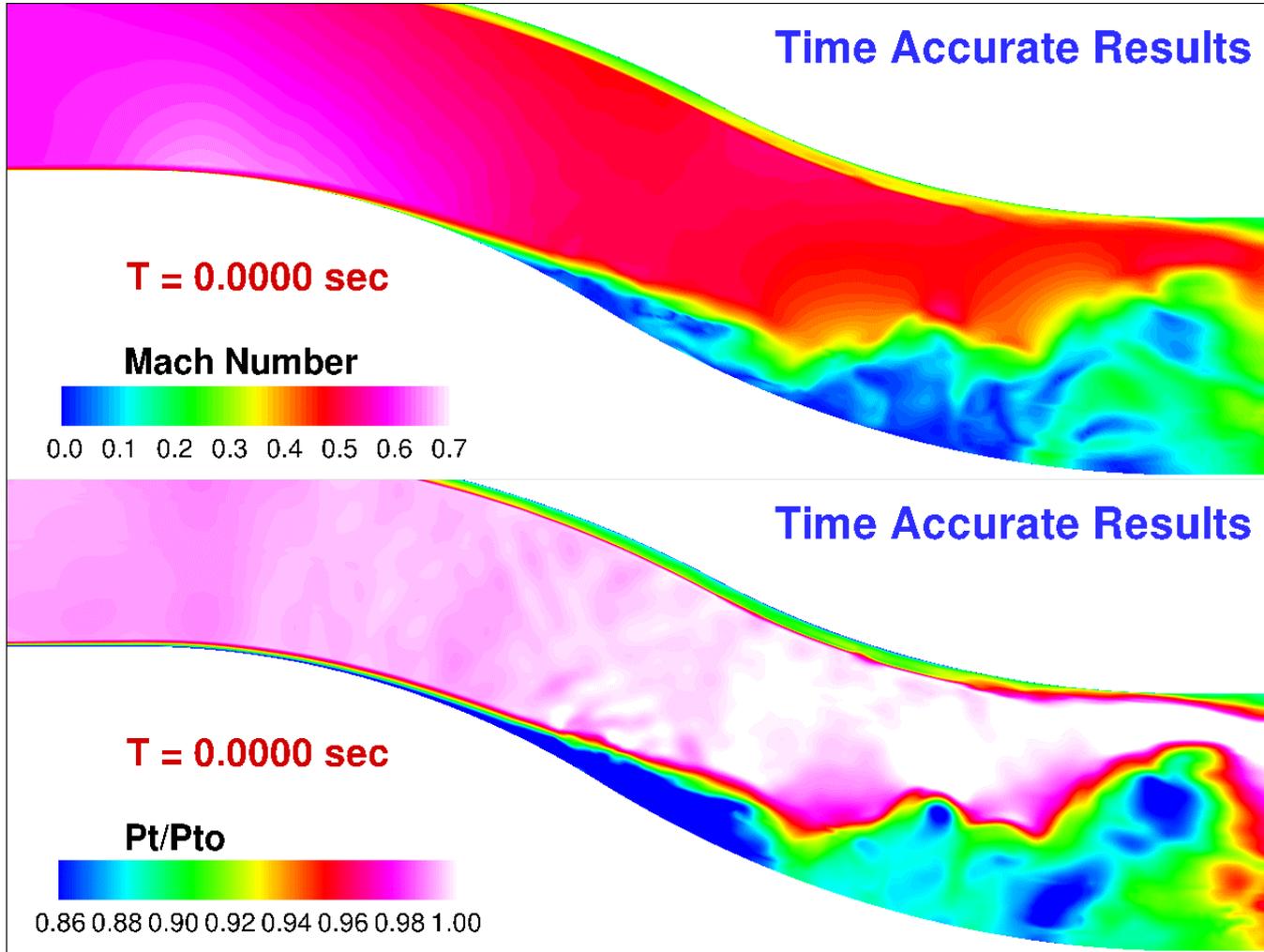
240mm²
P/P0=2.0



Time Accurate Run (URANS)

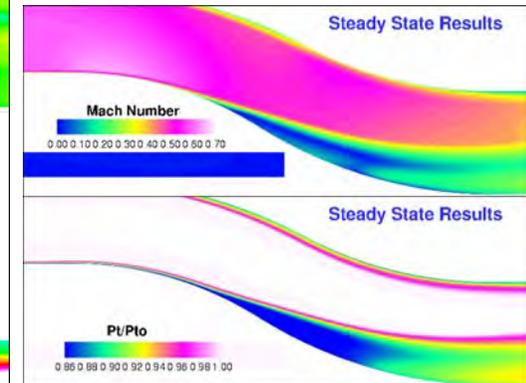
Time Accurate Results – ONERA S-Duct

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AIP Face Recovery
0.9707

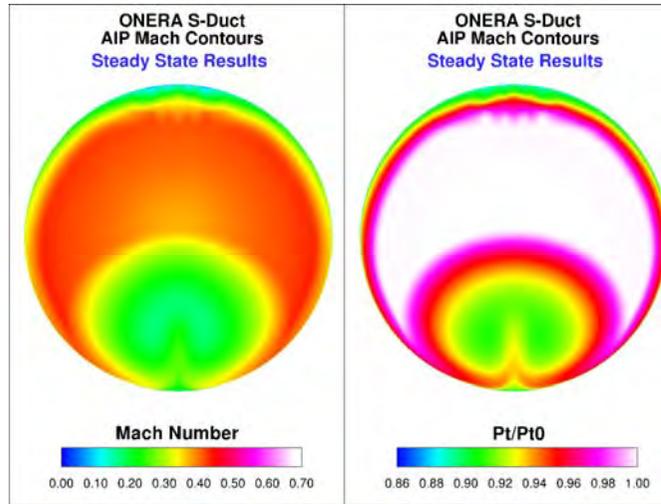
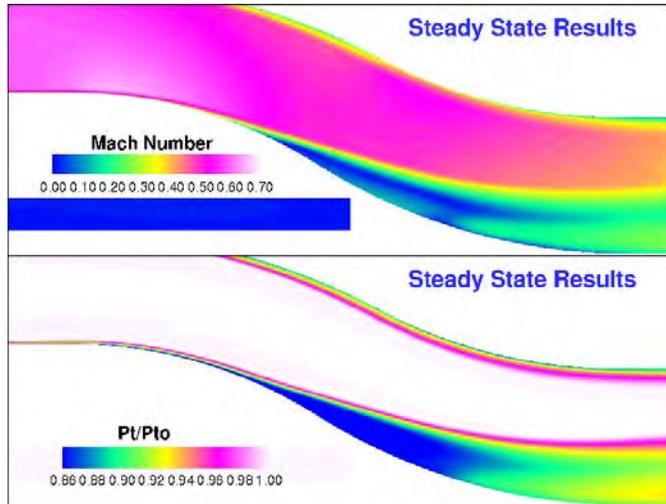
SAE DPCPAV
for all rings
0.0374



Time Accurate Results – ONERA S-Duct

Engineering, Operations & Technology

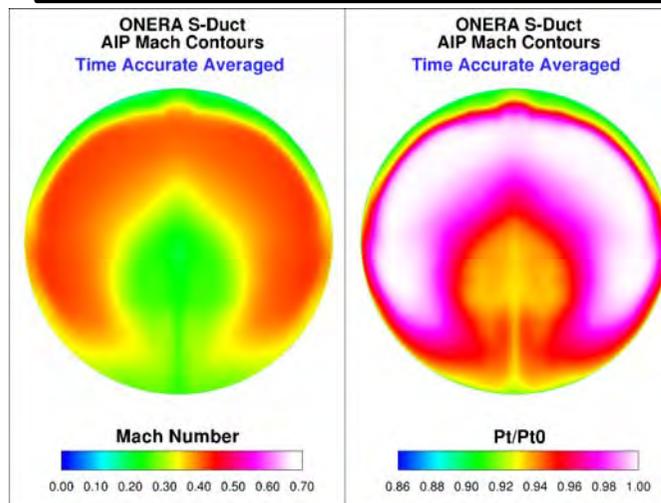
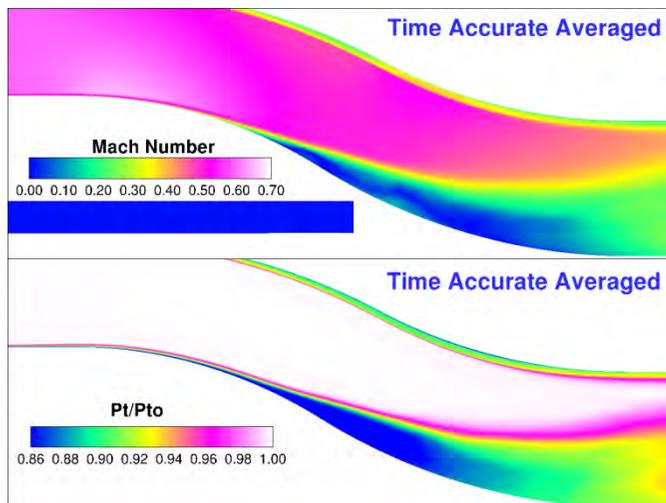
Steady State Results



AIP Face Recovery
0.9707

SAE DPCPAV
for all rings
0.0347

Time Accurate Averaged



AIP Face Recovery
0.9691

SAE DPCPAV
for all rings
0.0231

Conclusions

Engineering, Operations & Technology

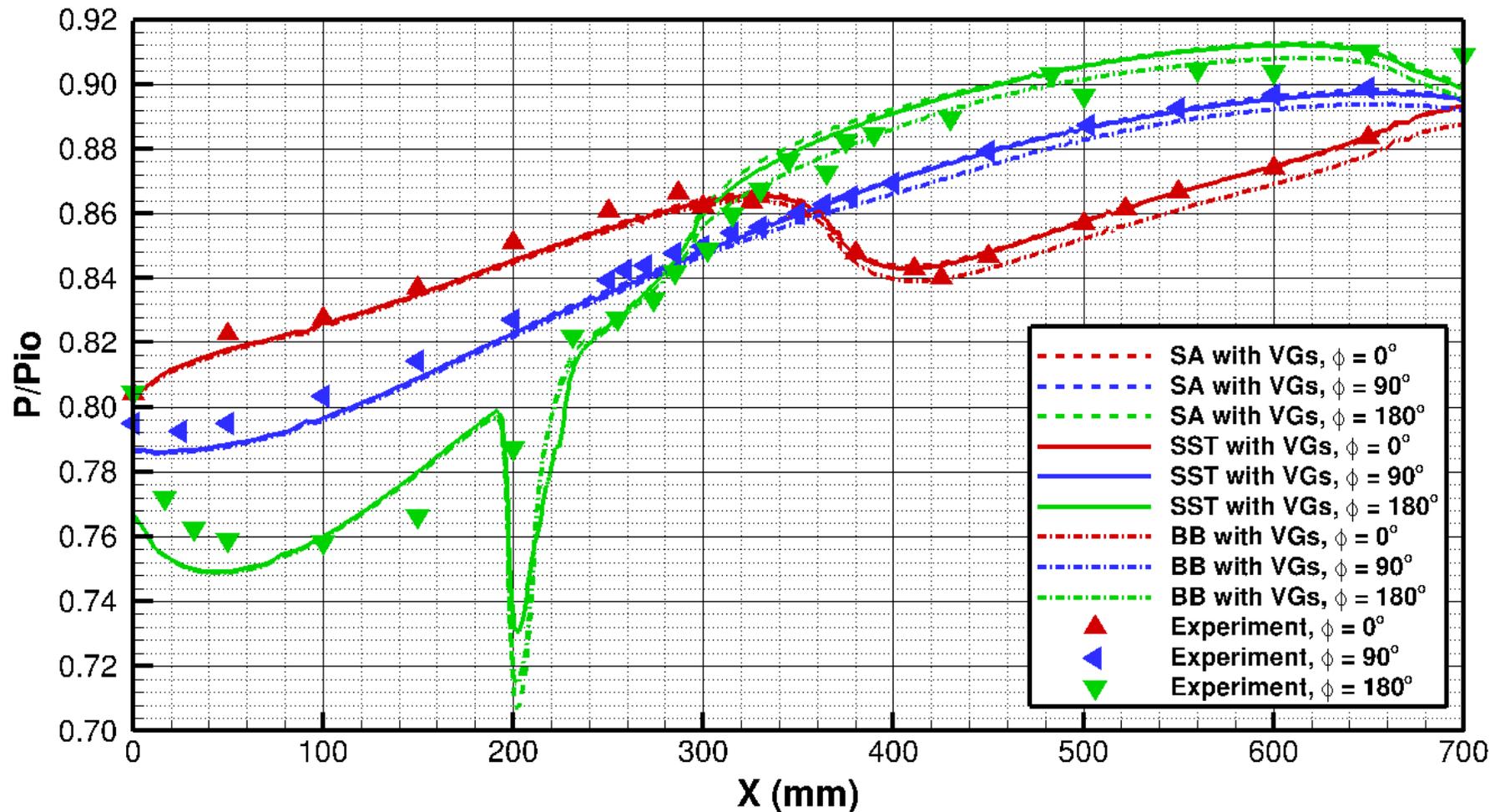
- CFD simulation using Overset grid approach is conducted for serpentine S-duct with and without VGs**
- CFD results are validated with experimental data**
- VG configurations include VG number, height, and orientation angle are simulated and studied**
- Recovery and circumferential distortion are computed to measure flow quality**
- CFD based AFC technique is employed to improve the flow quality**
- AFC offers better improvement relative to passive control**

Back Up Charts

Validation – Turbulence Model

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*Pressure Distribution along the S-duct at $\phi = 0^\circ, 90^\circ,$ and 180°
Overset Grid (OVERFLOW)
The Boeing Company - Turbulence Model Study*

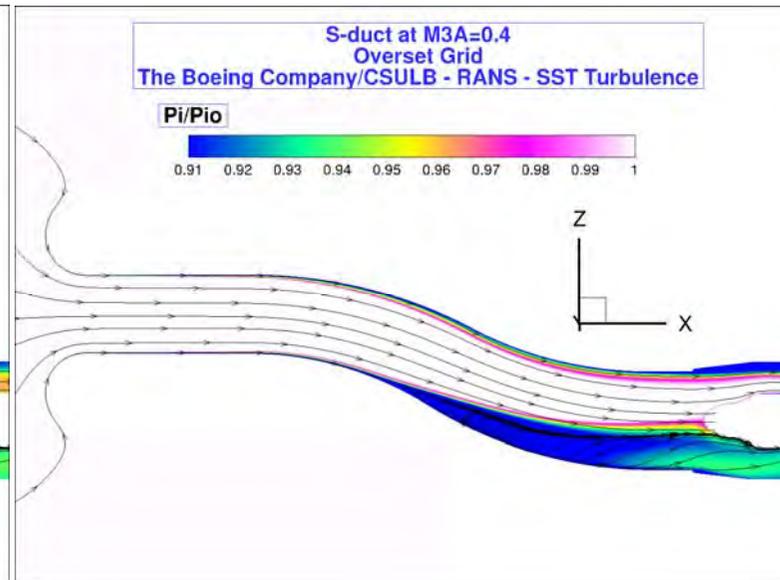
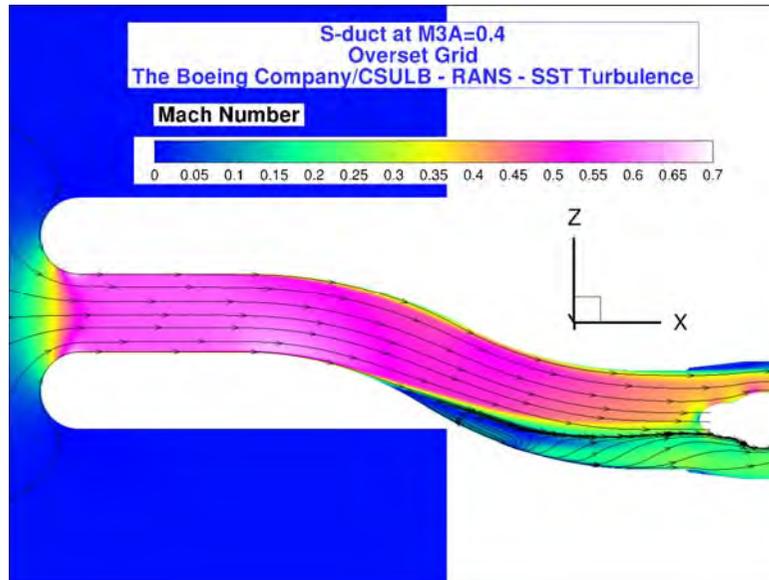


Results – Symmetric Plane Mach and Pi/Pio Contours

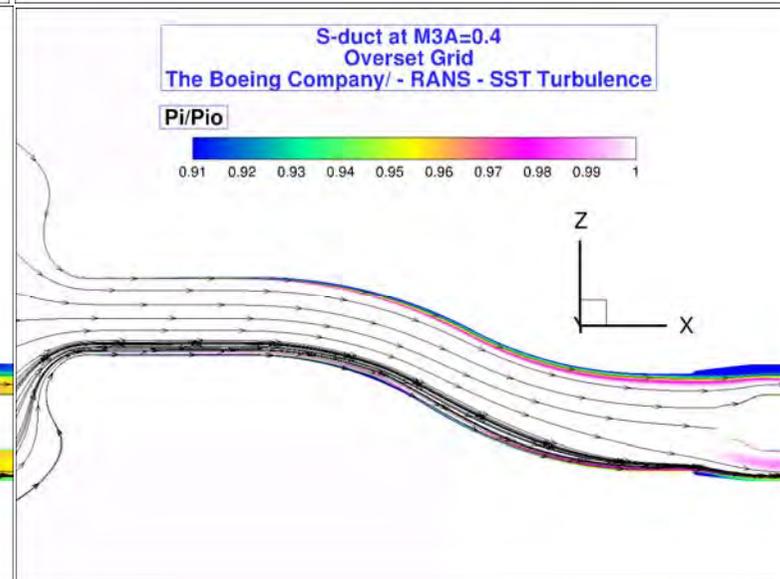
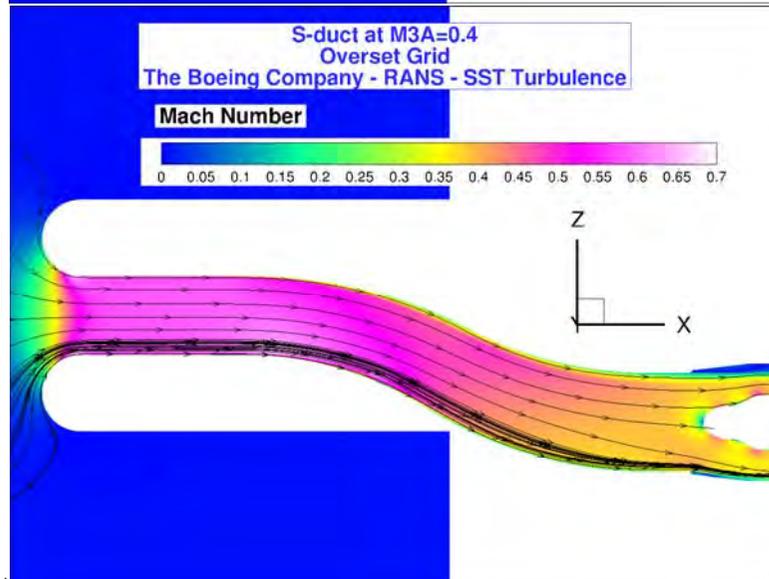
No VGs vs With VGs (mass flow = 2.427 kg/s)

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No VGs



With VGs

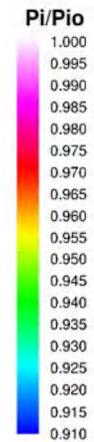
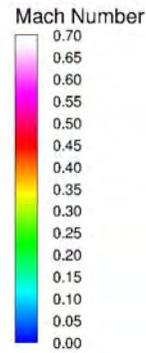
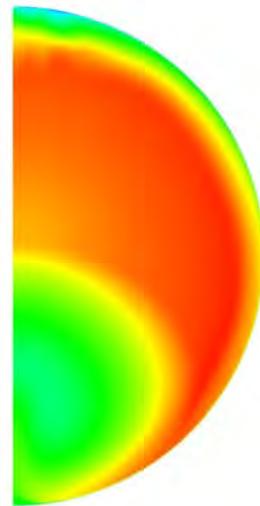


Results – AIP Mach and Pi/Pio Contours

No VGs vs With VGs (mass flow = 2.427 kg/s)

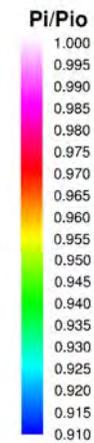
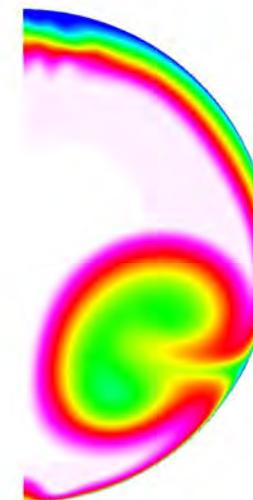
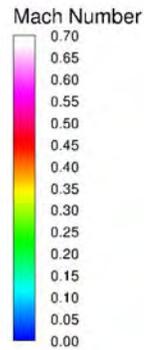
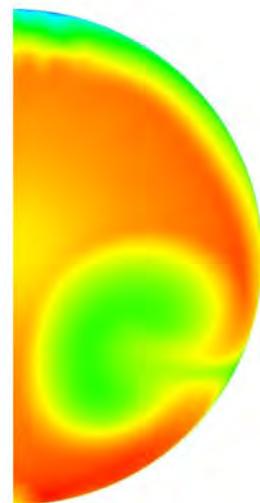
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No VGs



AIP

With VGs

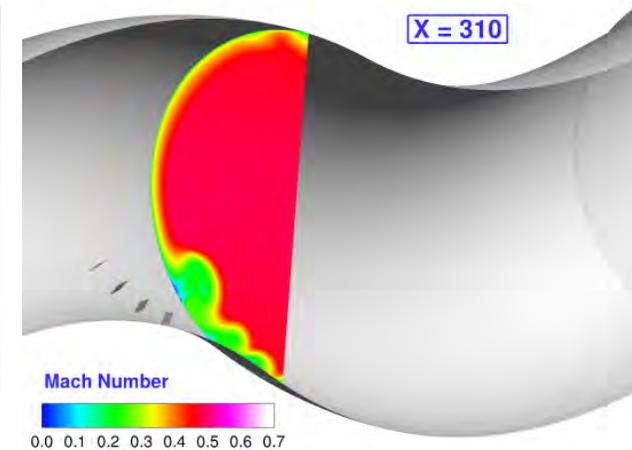
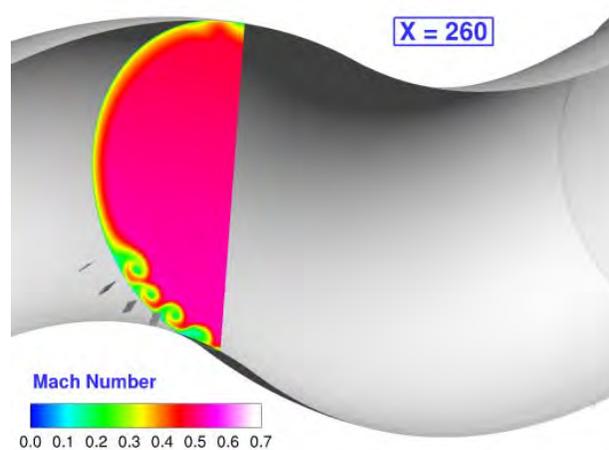
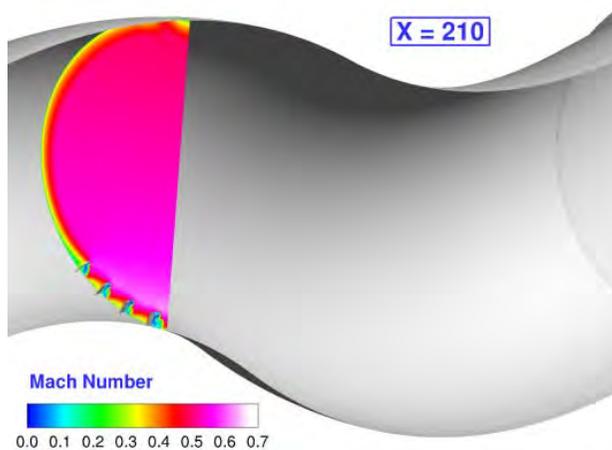
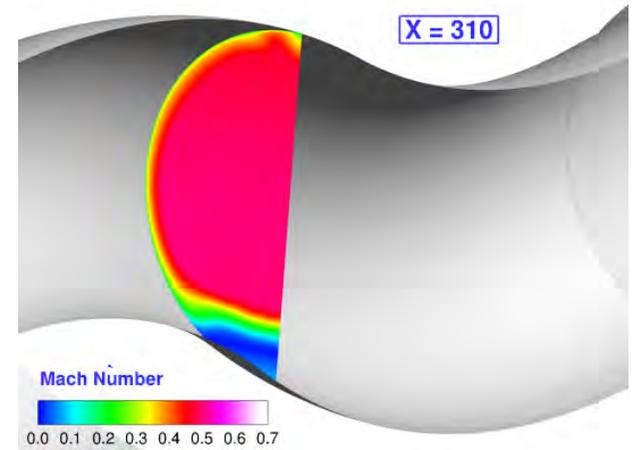
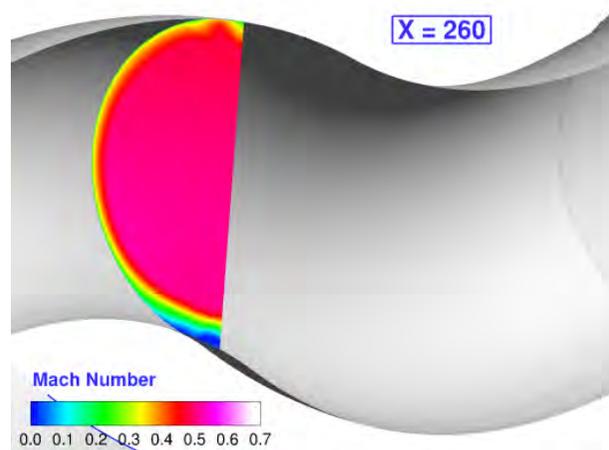
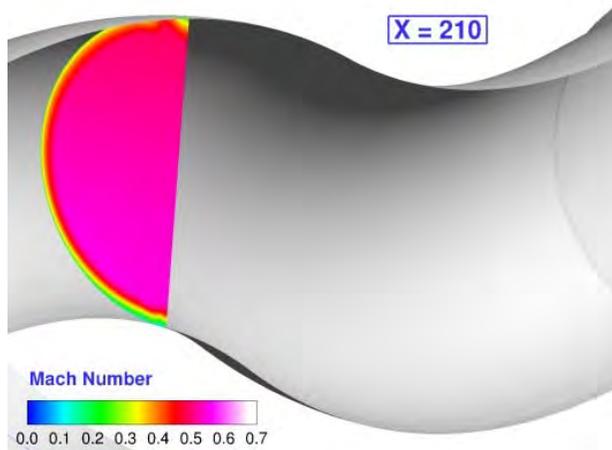


Results – Mach contours at different X cut

No VGs vs With VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology

No VGs



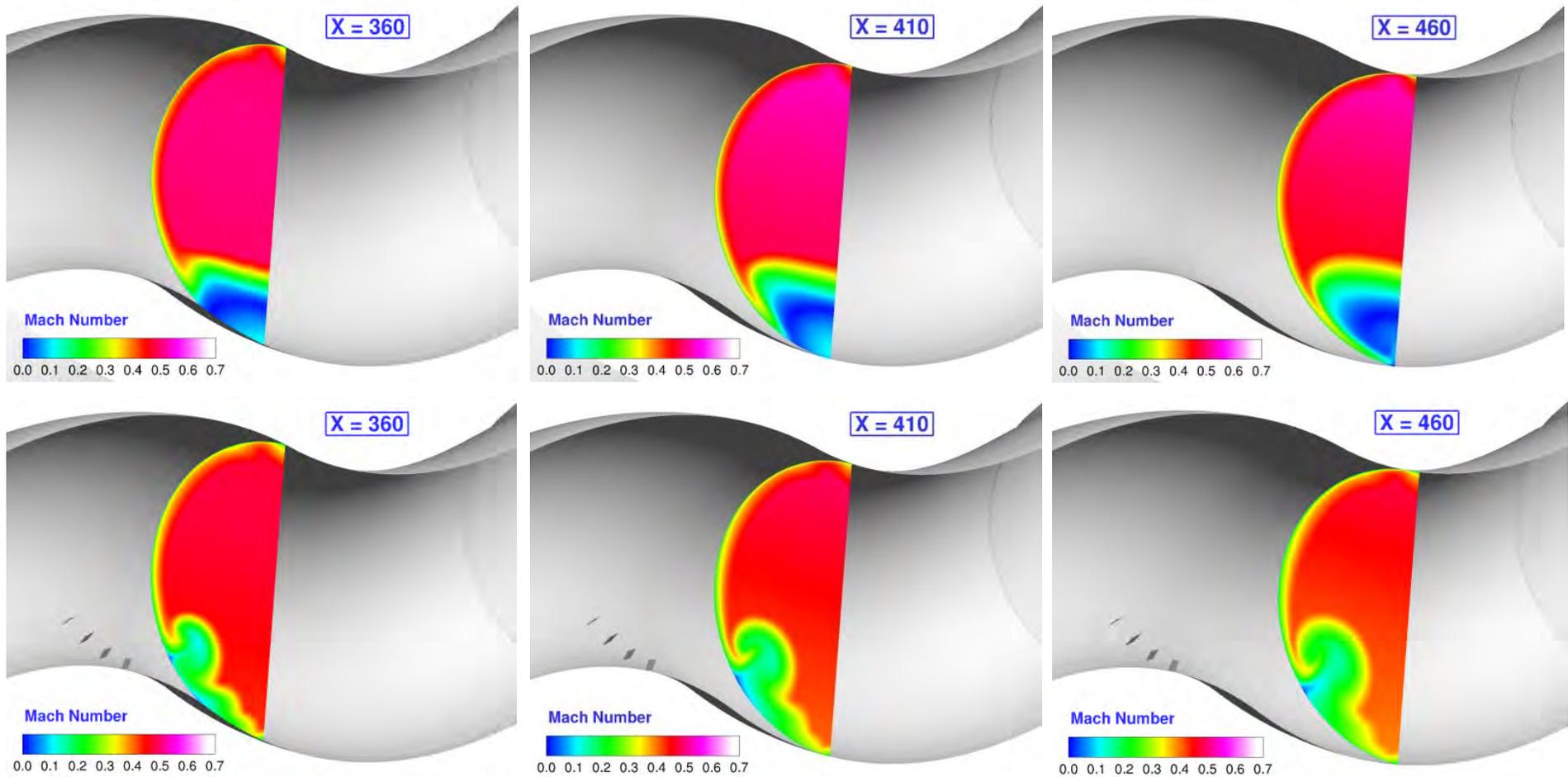
With VGs

Results – Mach contours at different X cut

No VGs vs With VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology

No VGs



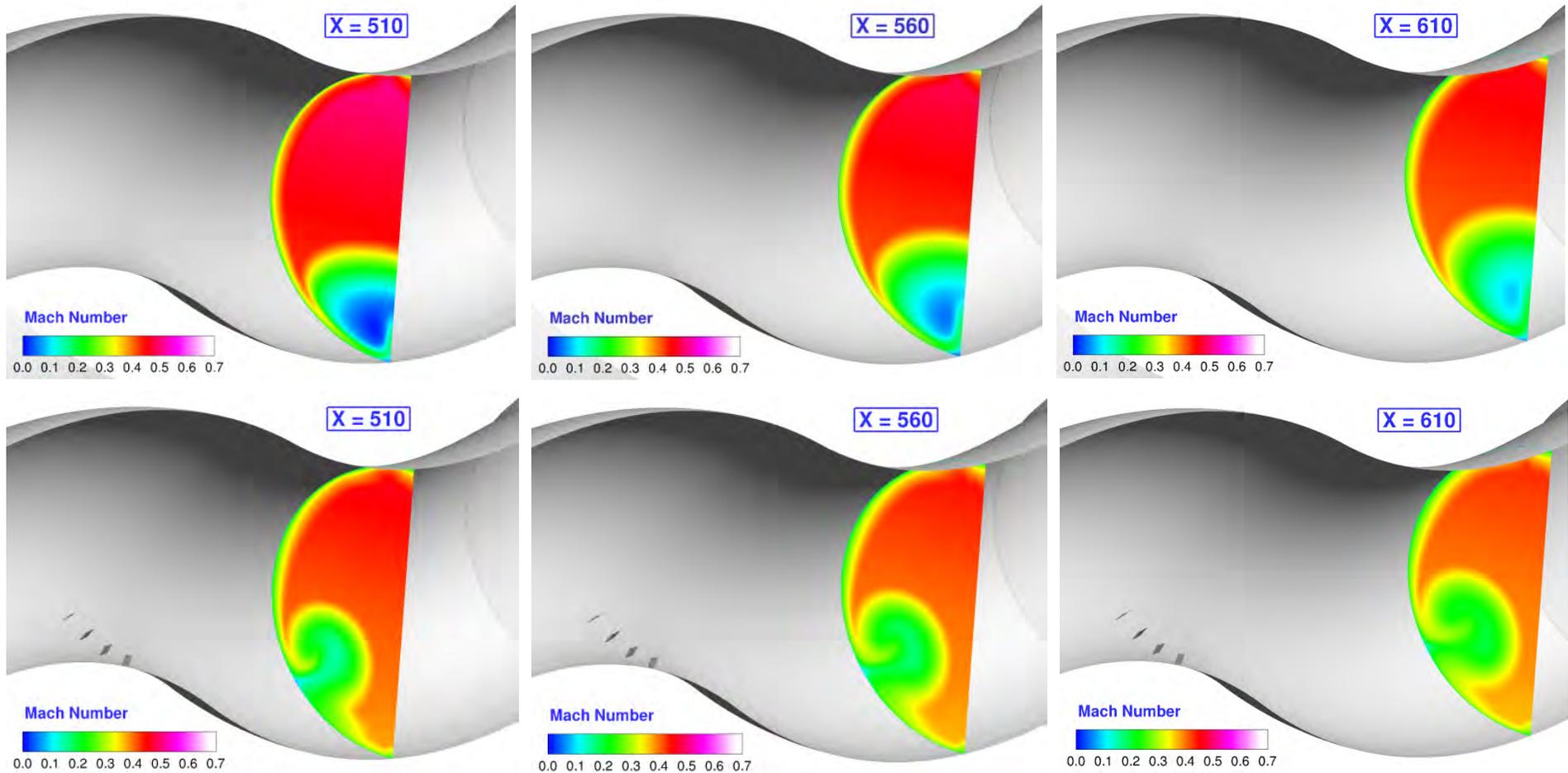
With VGs

Results – Mach contours at different X cut

No VGs vs With VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology

No VGs

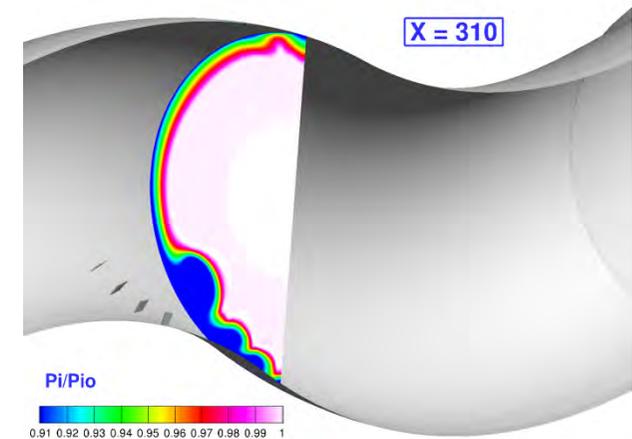
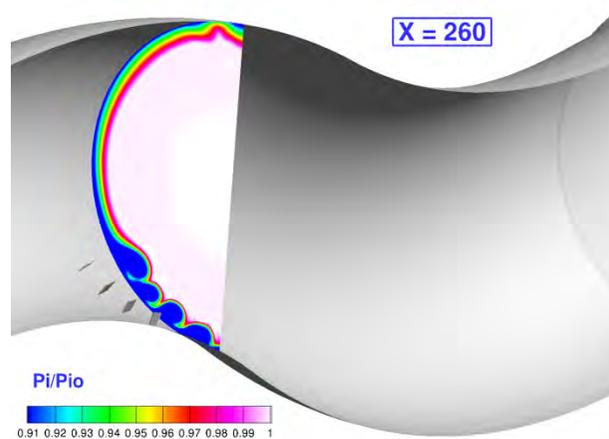
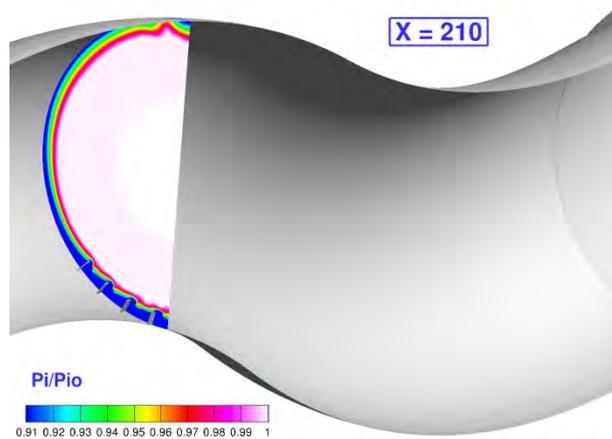
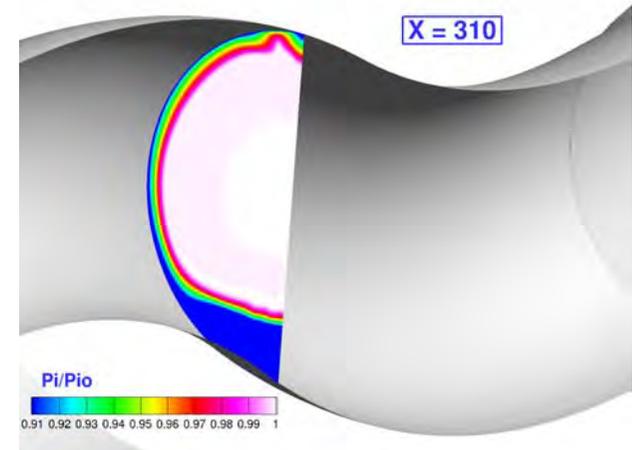
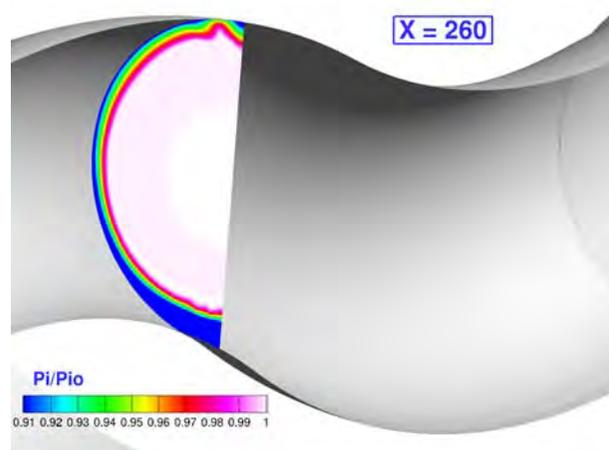
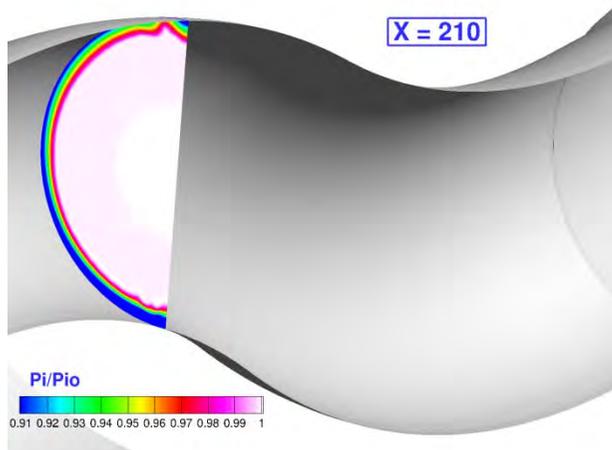


With VGs

Results – Pi/Pio Contours at different X cut No VGs vs With VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology

No VGs



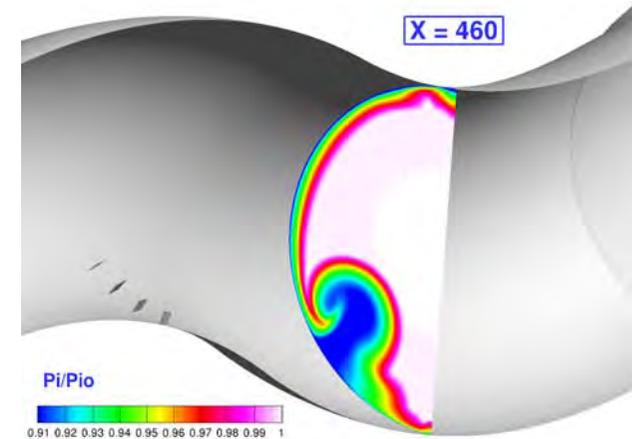
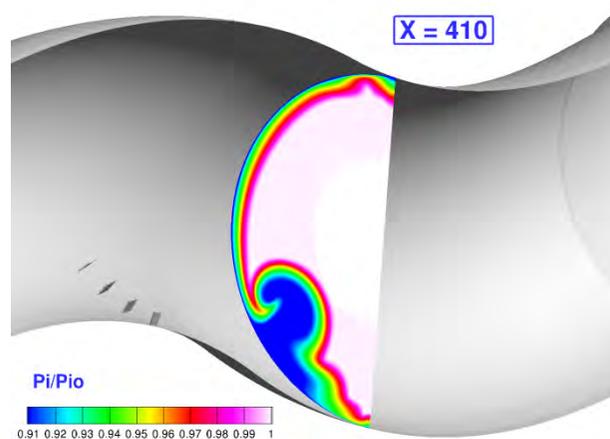
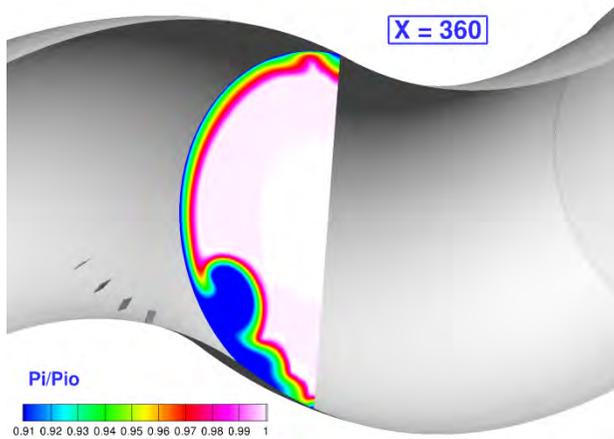
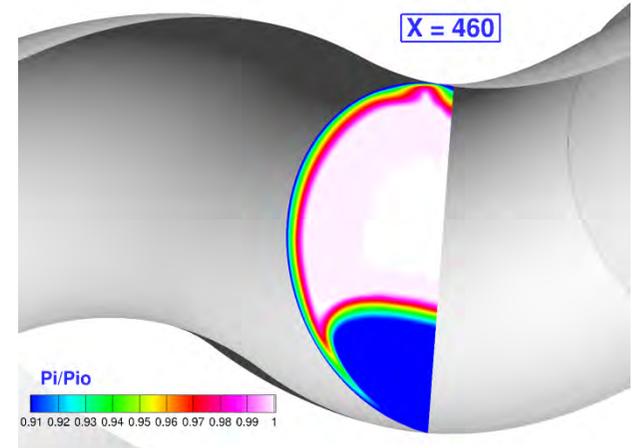
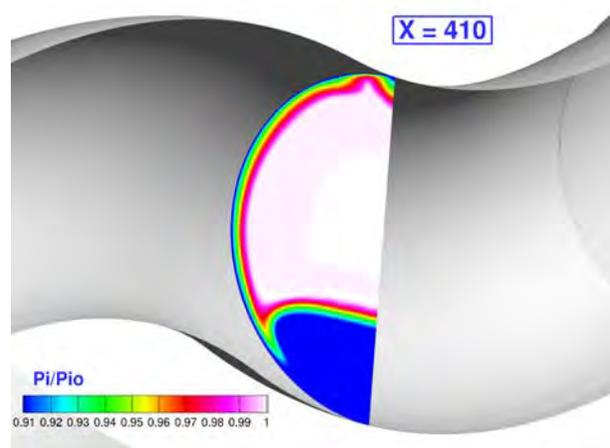
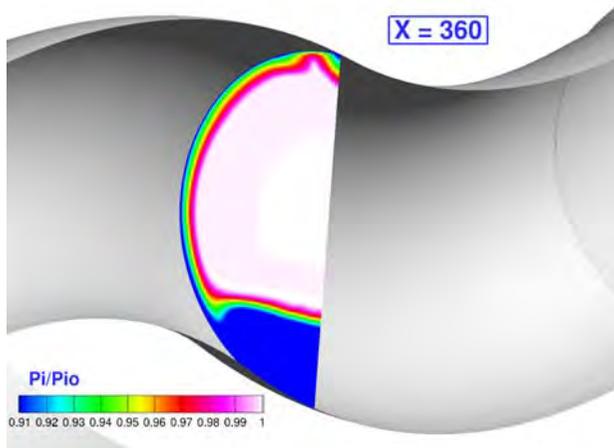
With VGs

Results – Pi/Pio Contours at different X cut

No VGs vs With VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology

No VGs

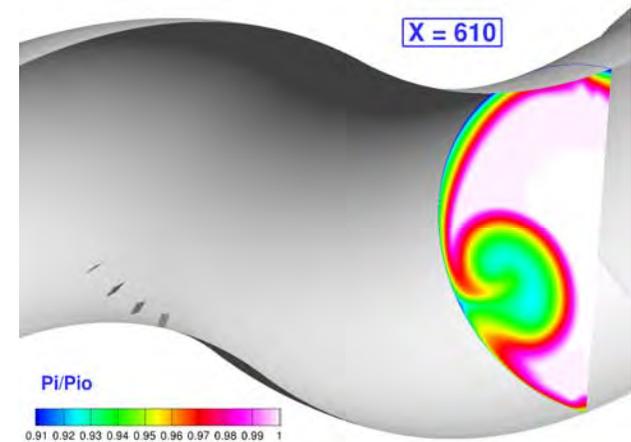
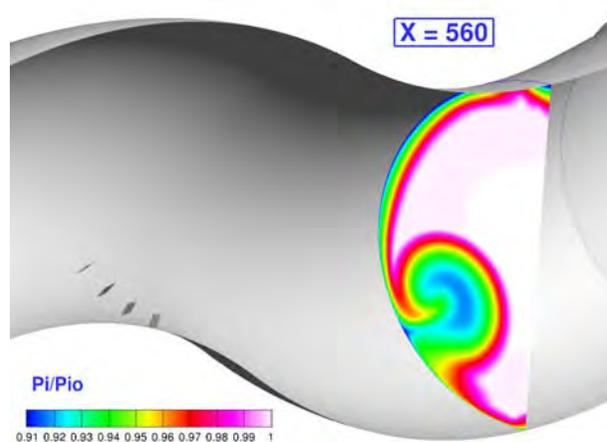
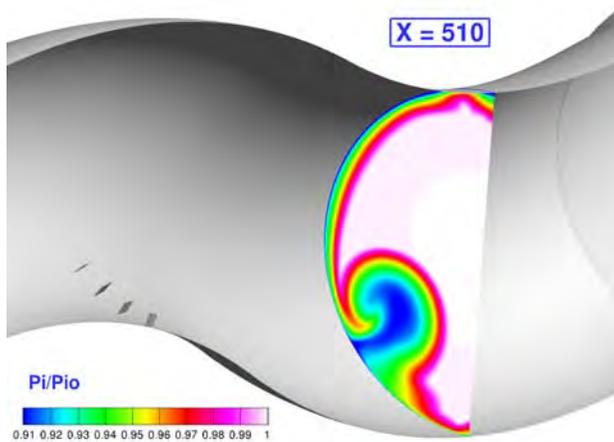
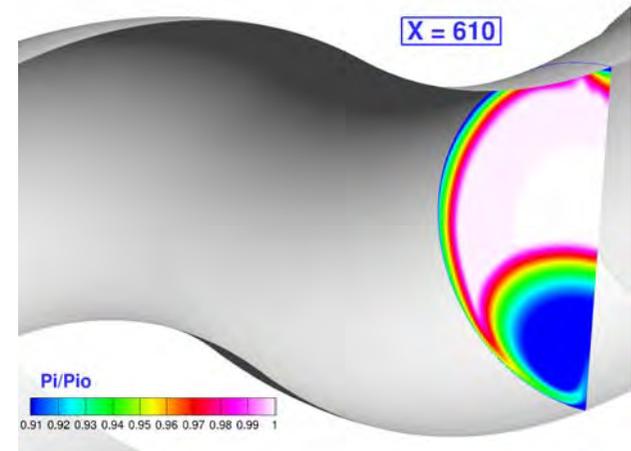
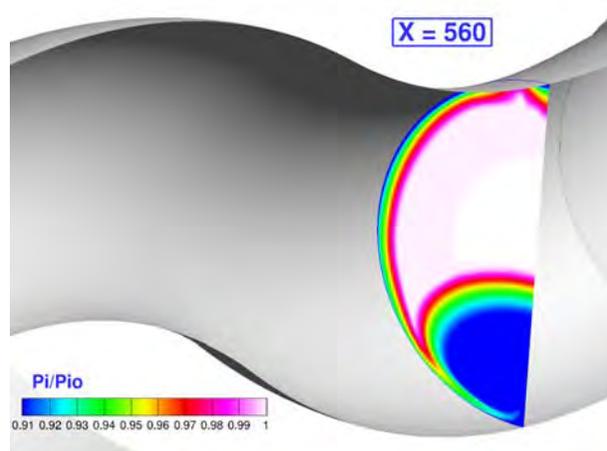
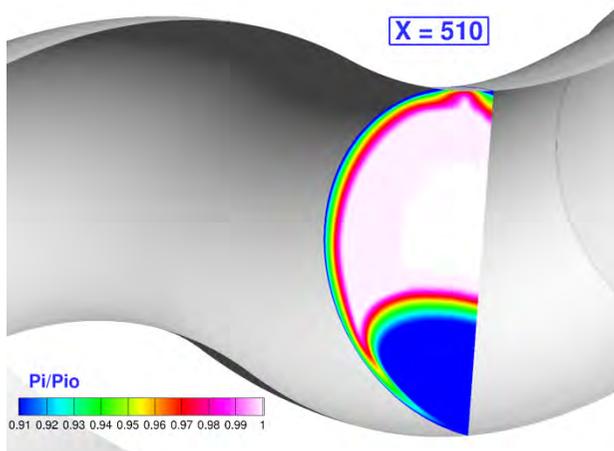


With VGs

Results – Pi/Pio Contours at different X cut No VGs vs With VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology

No VGs

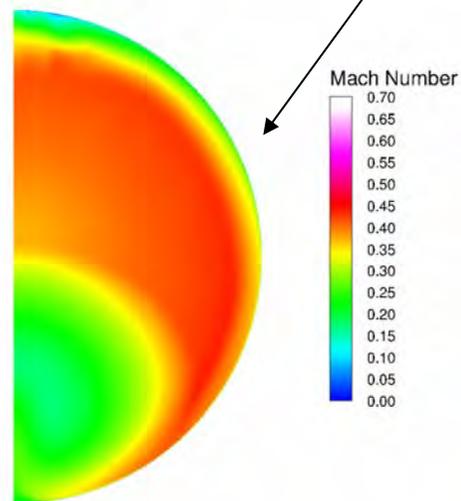
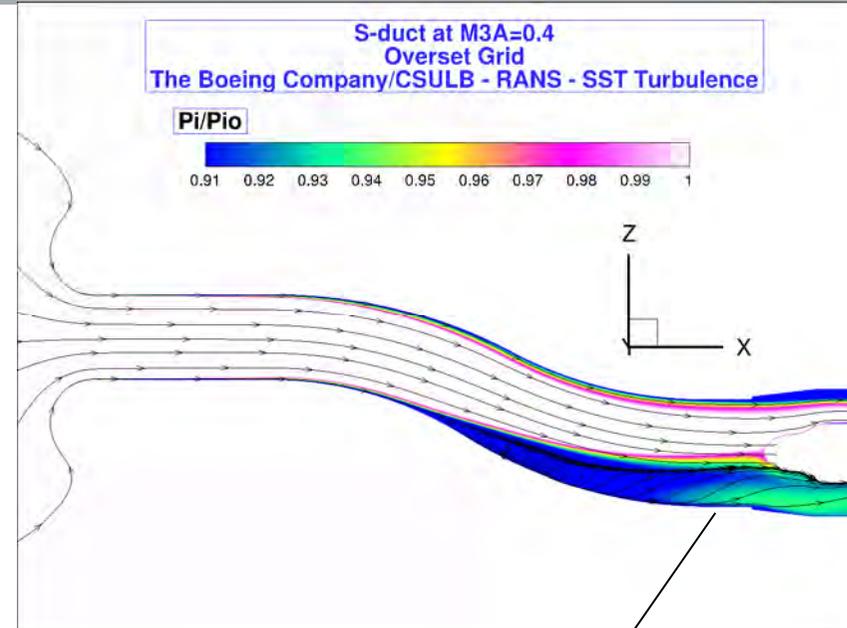
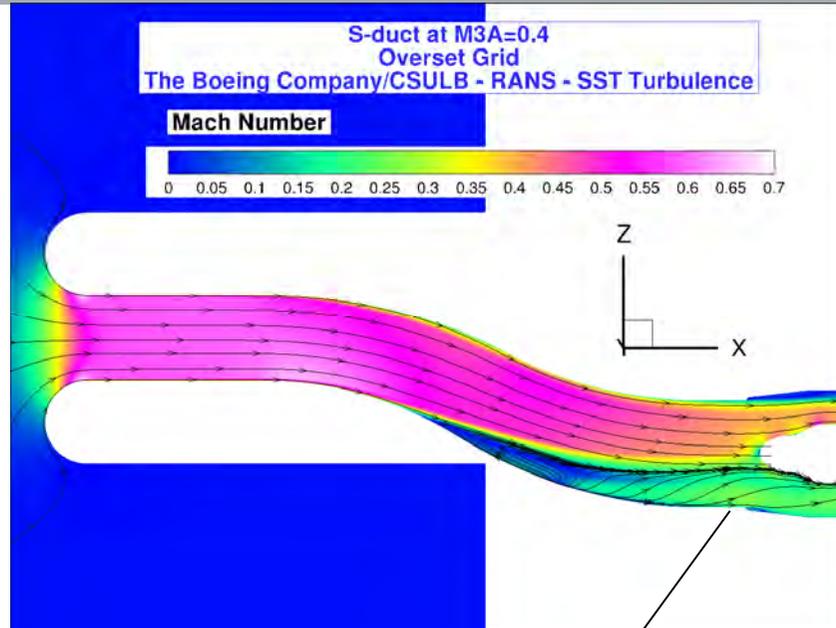


With VGs

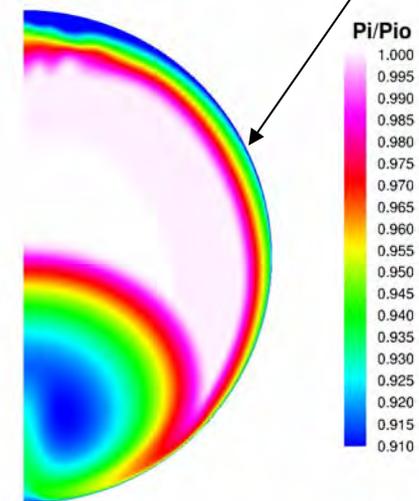
Results – Mach & Pi/Pio Contours

No VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology



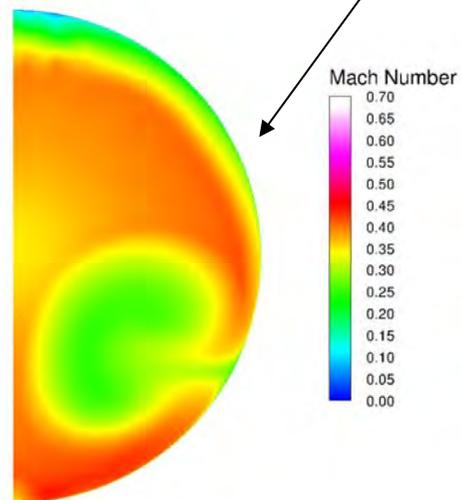
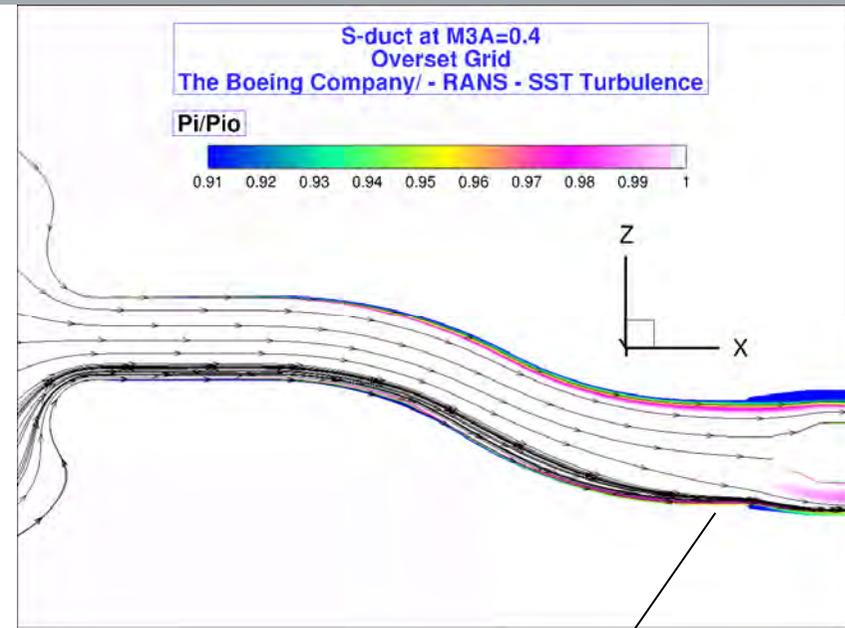
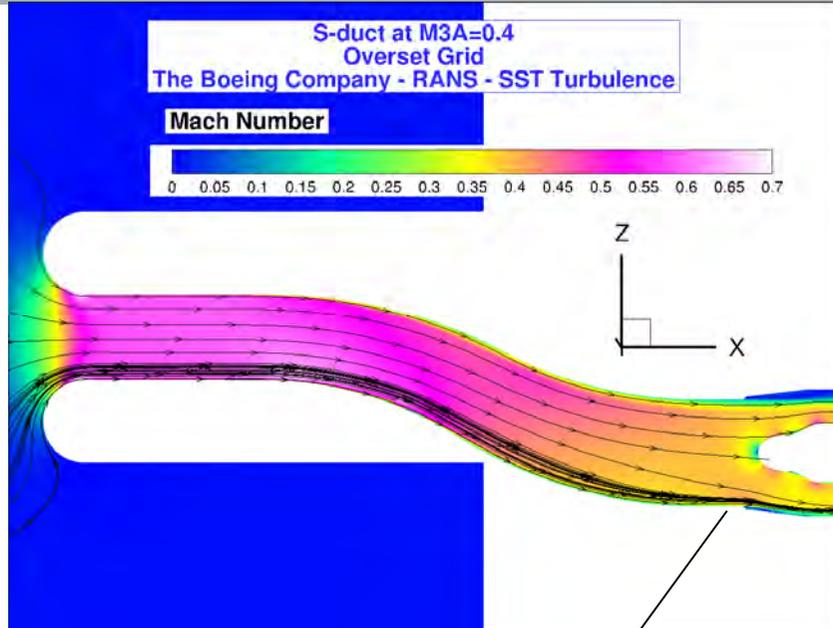
AIP plane



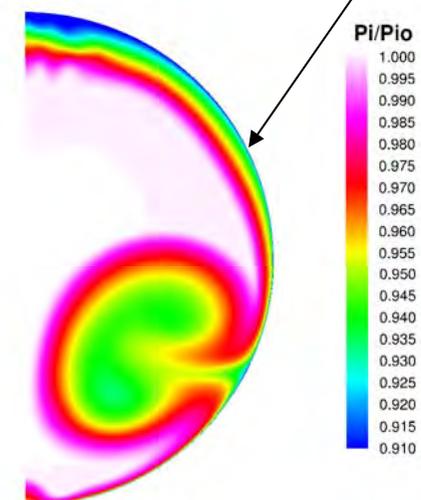
Results – Mach & Pi/Pio Contours

With VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology



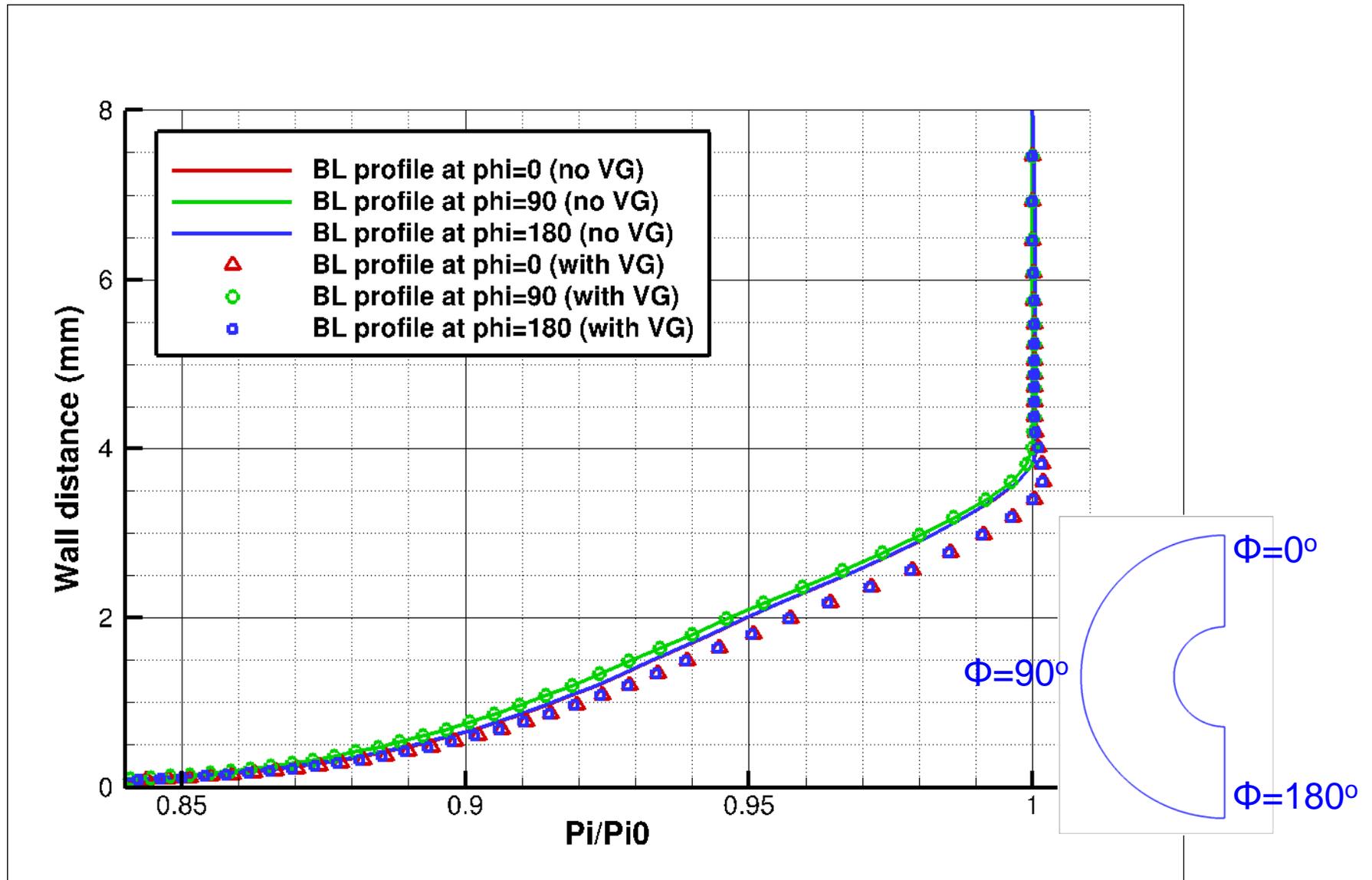
AIP



Results – BL Profiles (x=-76.58 mm)

VGs vs No VGs (mass flow = 2.427 kg/s)

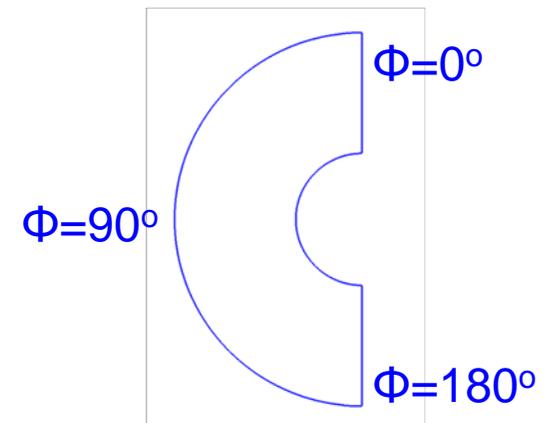
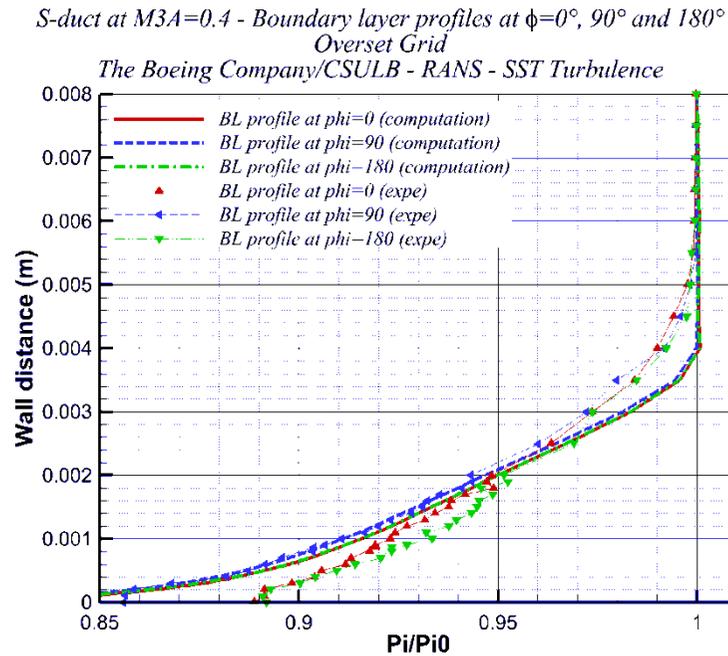
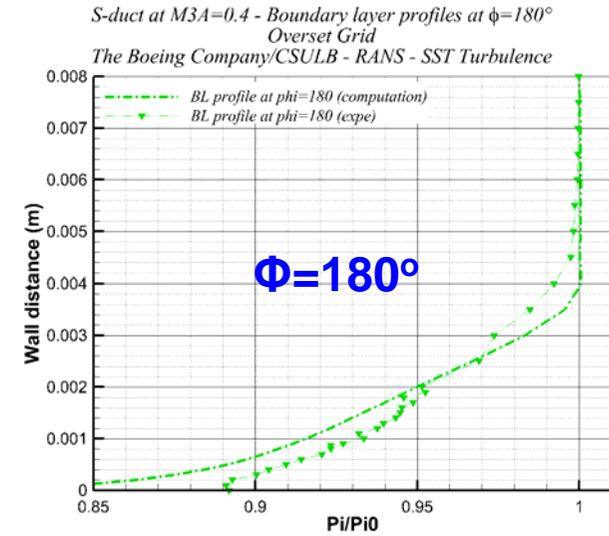
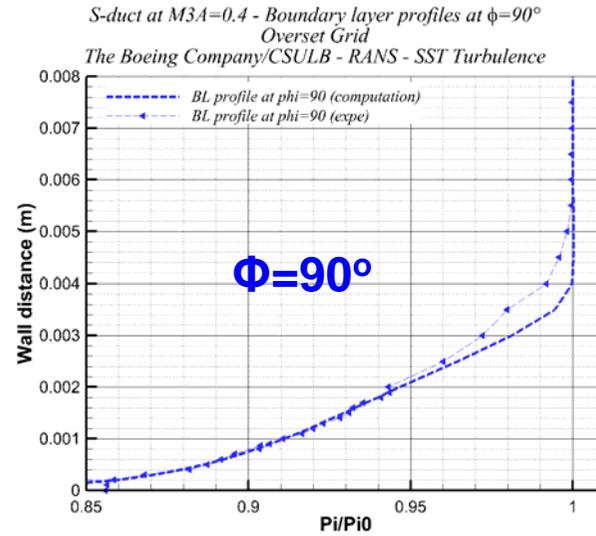
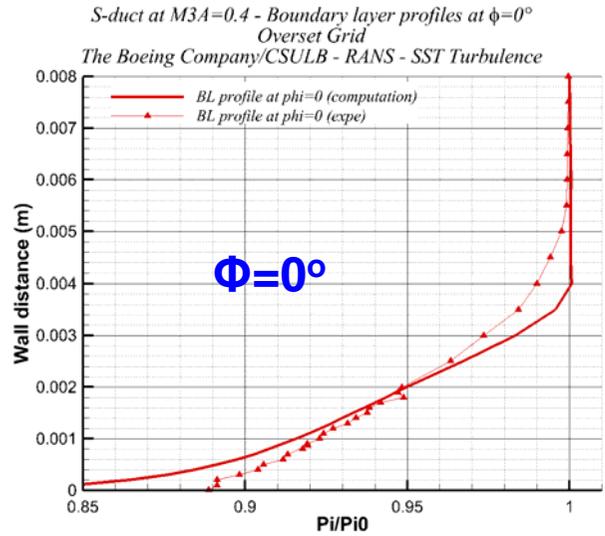
Engineering, Operations & Technology



Results – BL Profiles (x=-76.58 mm)

No VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology

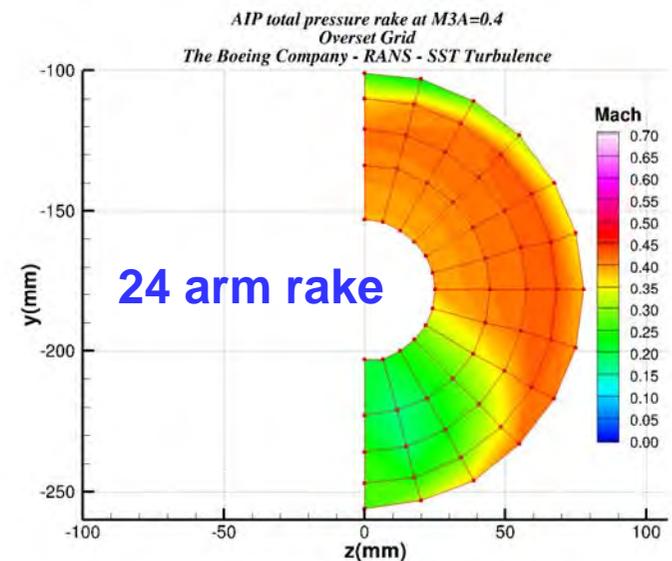
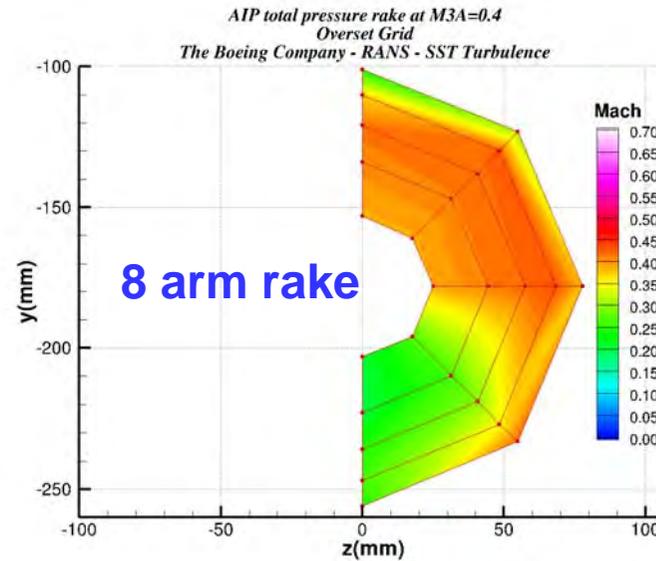
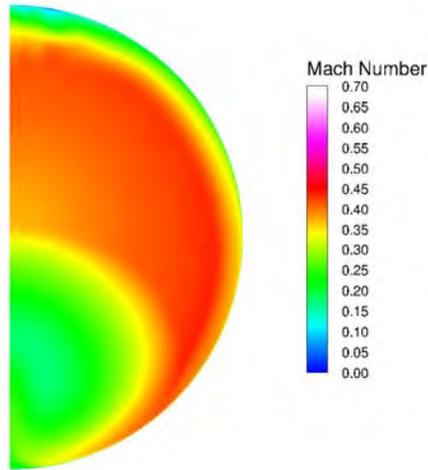


Results – AIP Mach Contours

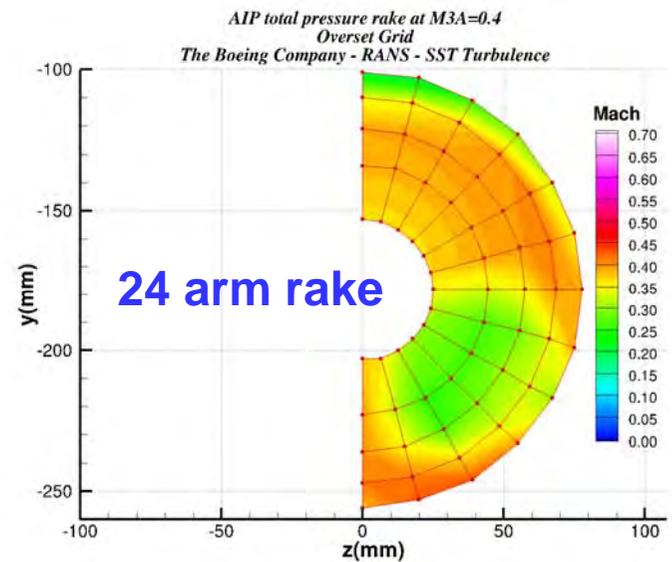
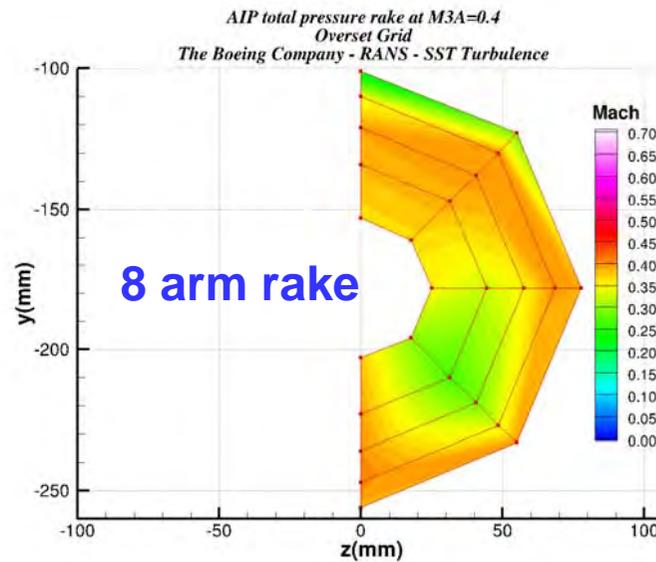
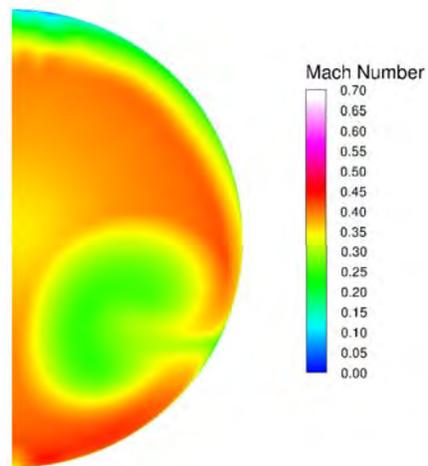
No VGs vs With VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology

No VGs



With VGs

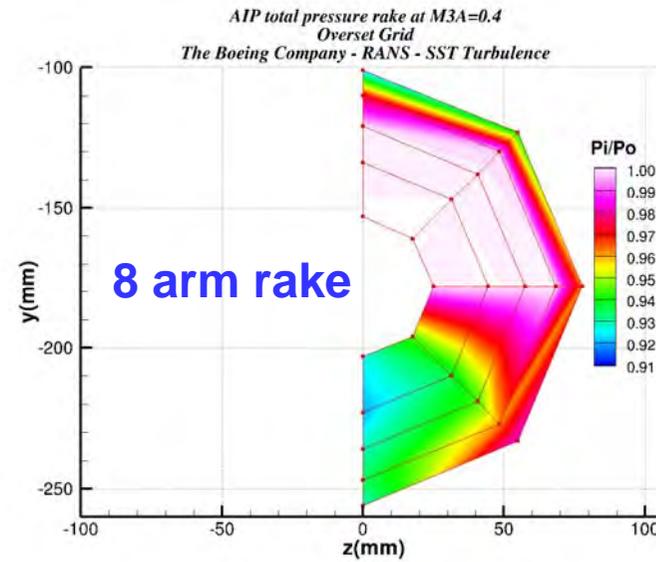
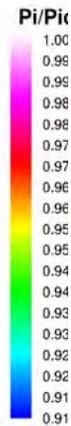


Results – AIP P_i/P_{i0} Contours

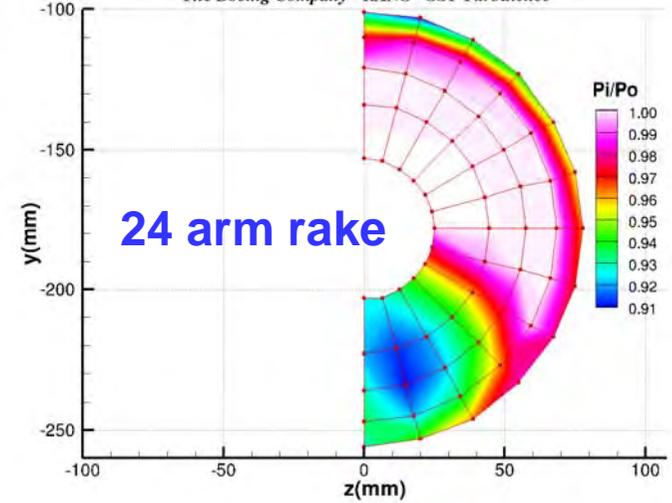
No VGs vs With VGs (mass flow = 2.427 kg/s)

Engineering, Operations & Technology

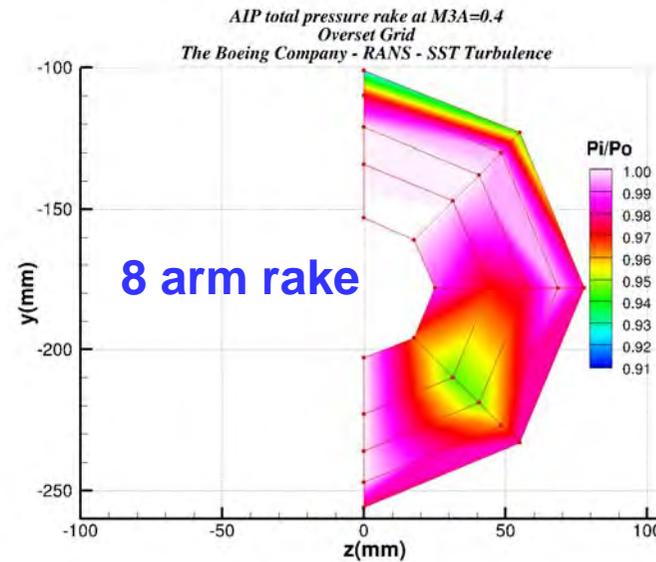
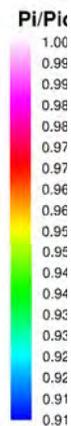
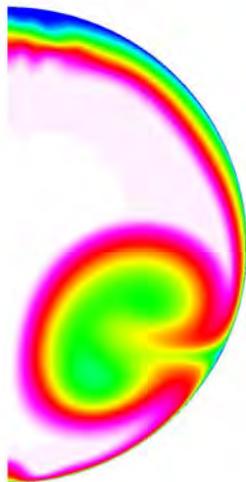
No VGs



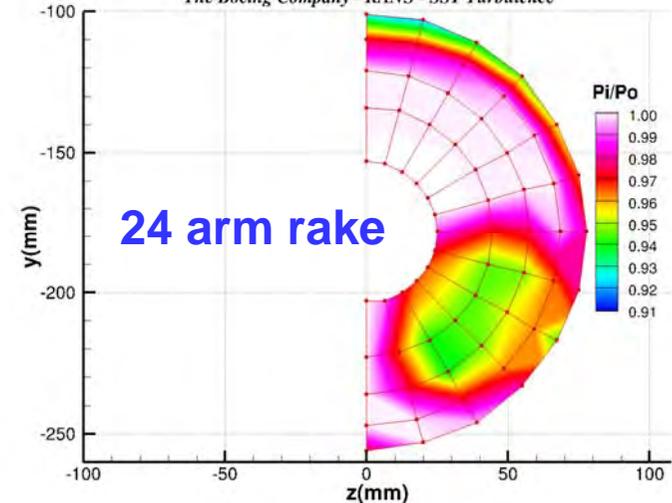
AIP total pressure rake at $M3A=0.4$
Overset Grid
The Boeing Company - RANS - SST Turbulence



With VGs

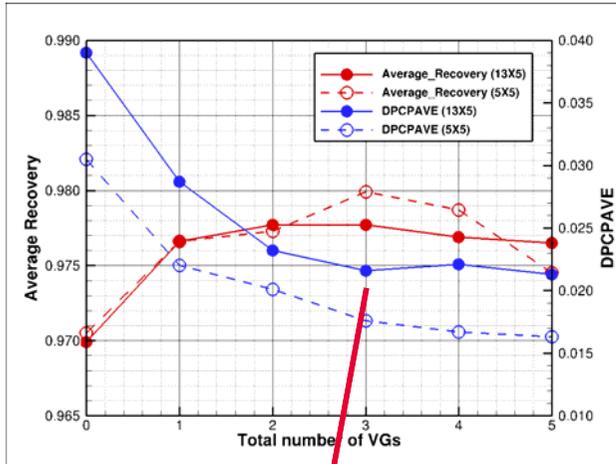


AIP total pressure rake at $M3A=0.4$
Overset Grid
The Boeing Company - RANS - SST Turbulence

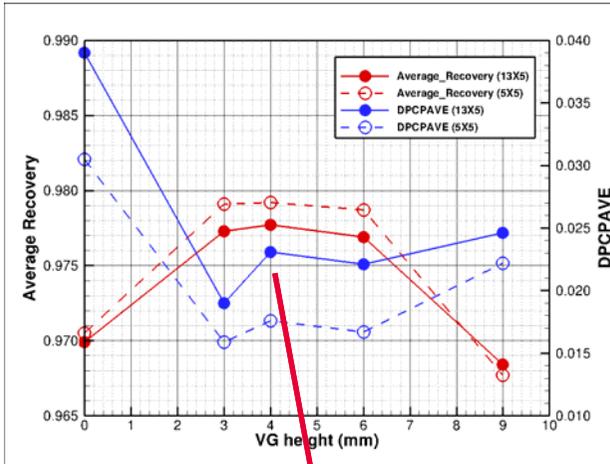
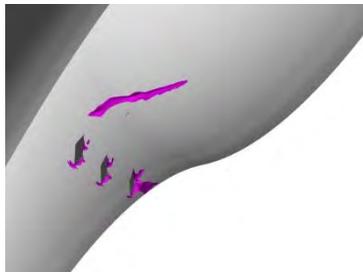
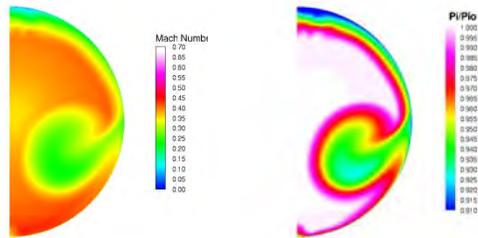


Passive Flow Control - VGs

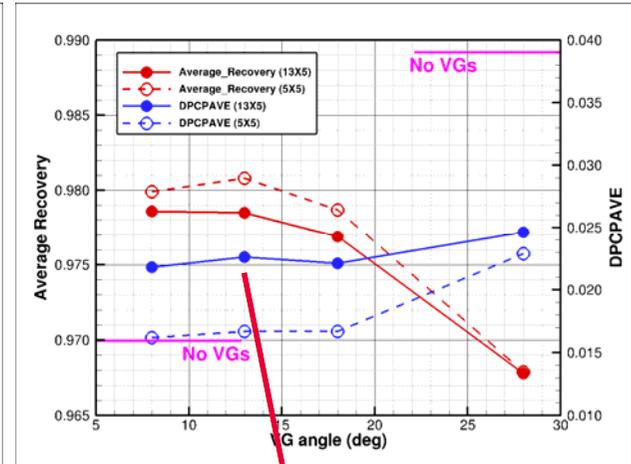
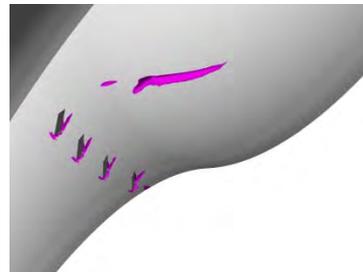
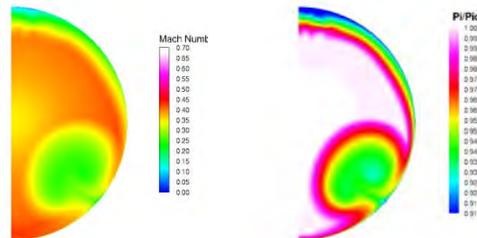
Engineering, Operations & Technology



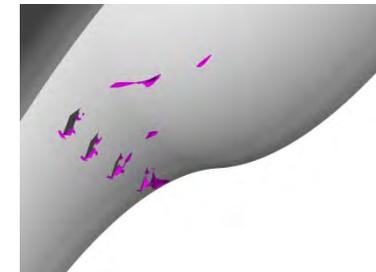
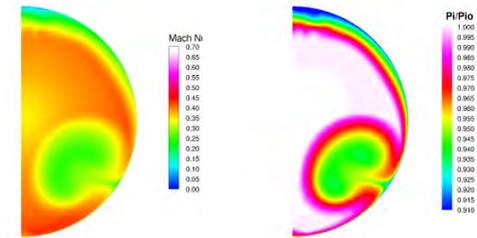
3 VGs, 6mm height, 18 deg



4 VGs, 4mm height, 18 deg

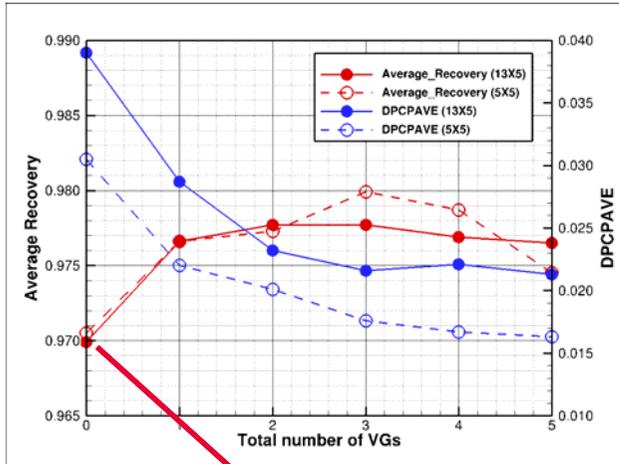


4 VGs, 6mm height, 13 deg

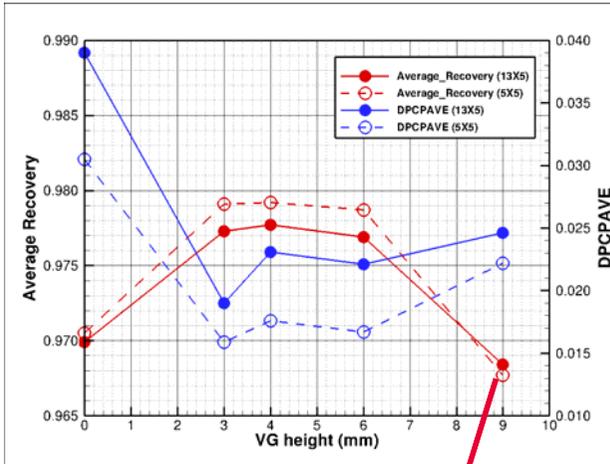
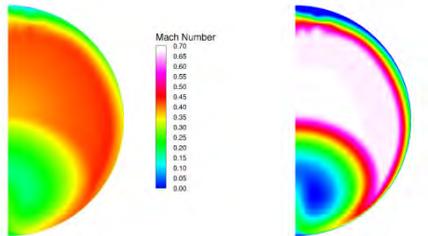


Passive Flow Control - VGs

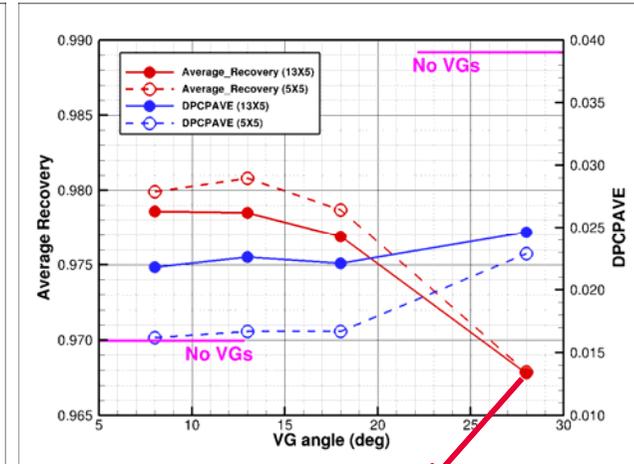
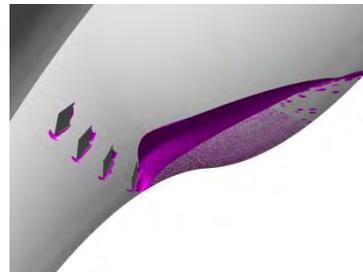
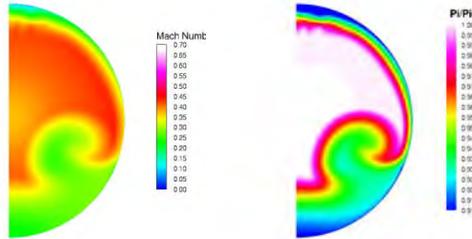
Engineering, Operations & Technology



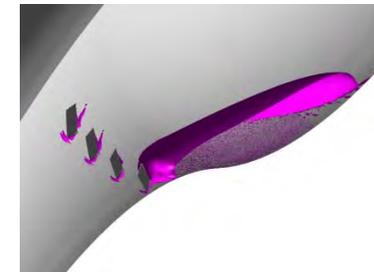
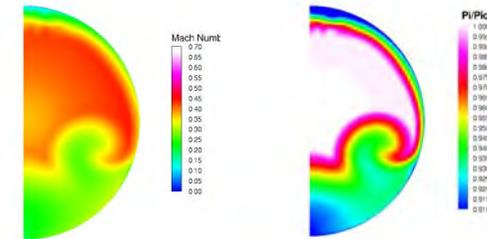
No VGs,



4 VGs, 9mm height, 18 deg



4 VGs, 6mm height, 28 deg

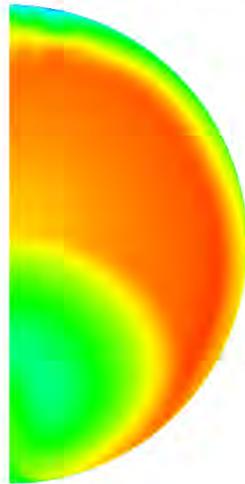


Results – Mach & Pi/Pio Contours

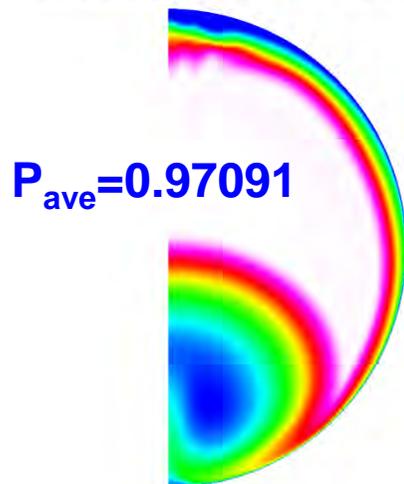
Standard Case (mass flow = 2.427 kg/s)

Engineering, Operations & Technology

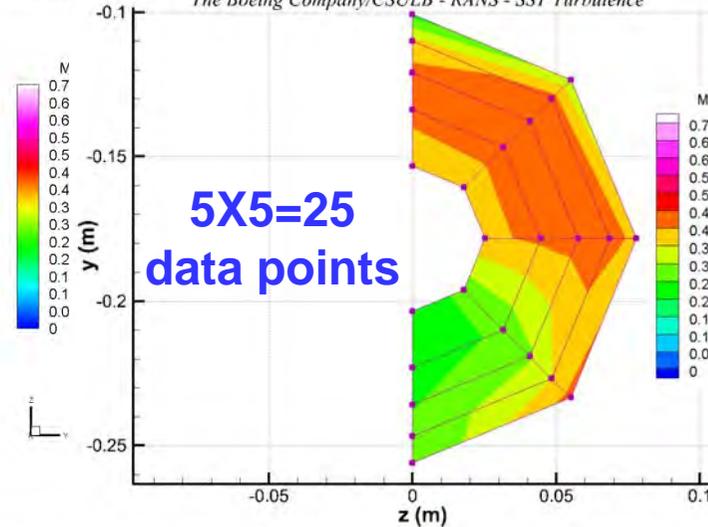
S-duct at $M3A=0.4$
Overset Grid
The Boeing Company/CSULB - RANS - SST Turbulence



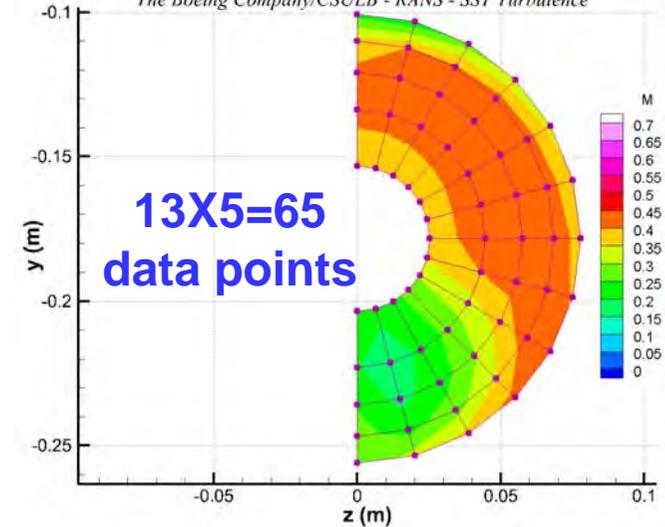
S-duct at $M3A=0.4$
Overset Grid
The Boeing Company/CSULB - RANS - SST Turbulence



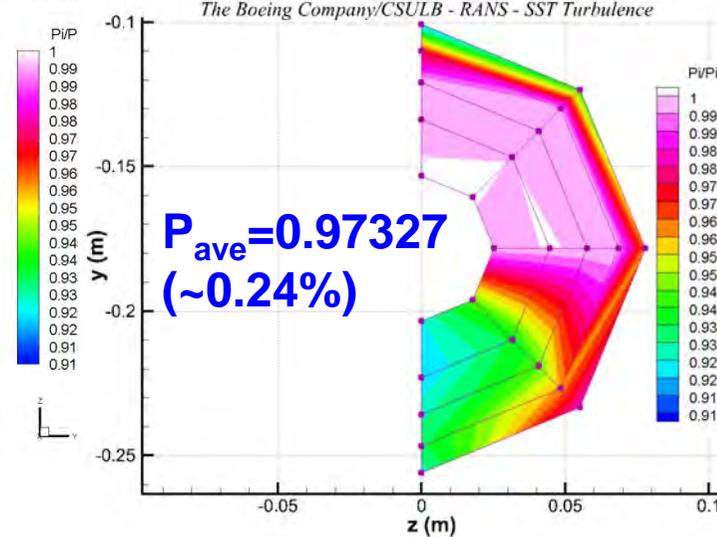
AIP 40 probe unsteady total pressure rake at $M3A=0.4$
Overset Grid
The Boeing Company/CSULB - RANS - SST Turbulence



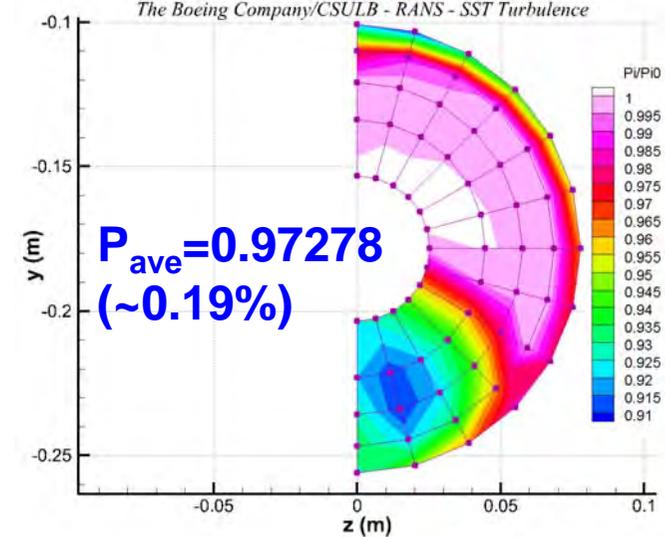
AIP Virtual total pressure rake at $M3A=0.4$
Overset Grid
The Boeing Company/CSULB - RANS - SST Turbulence



AIP 40 probe unsteady total pressure rake at $M3A=0.4$
Overset Grid
The Boeing Company/CSULB - RANS - SST Turbulence

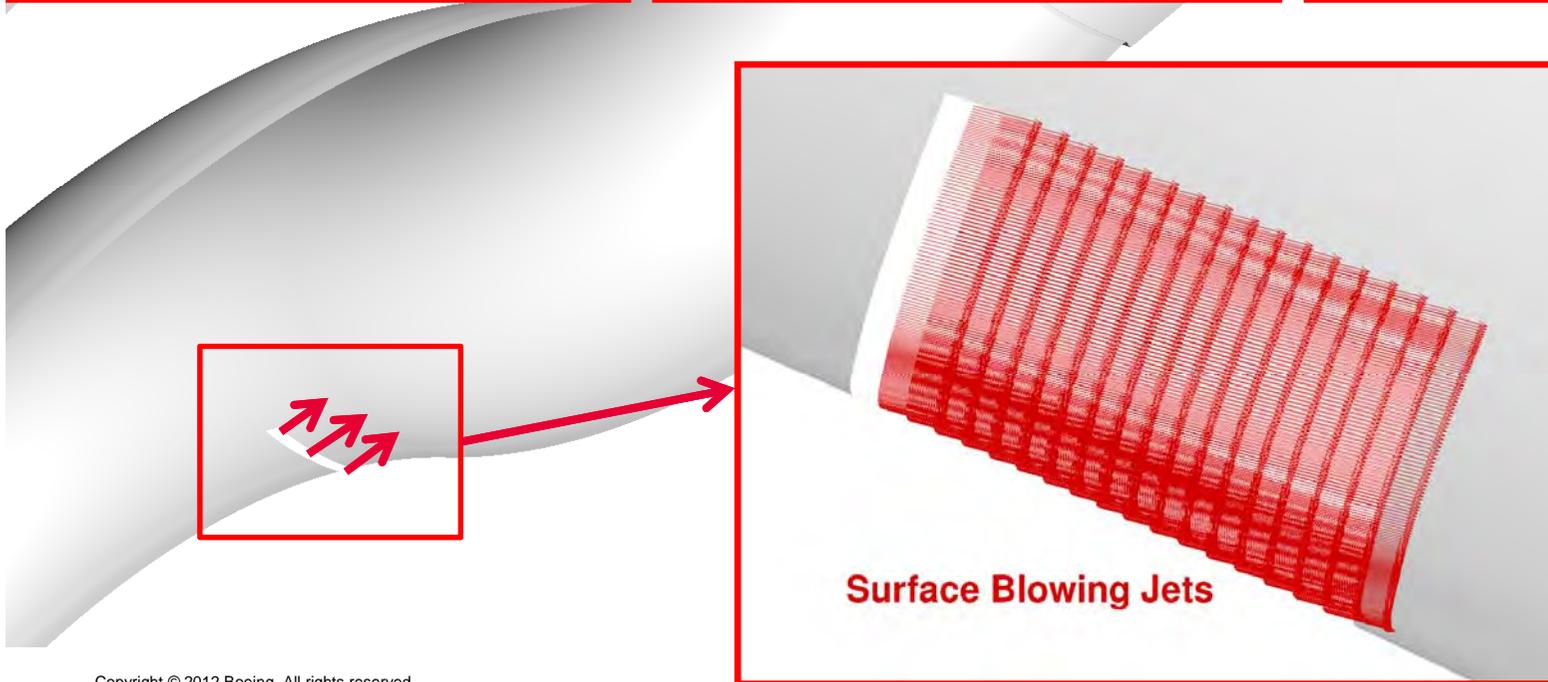
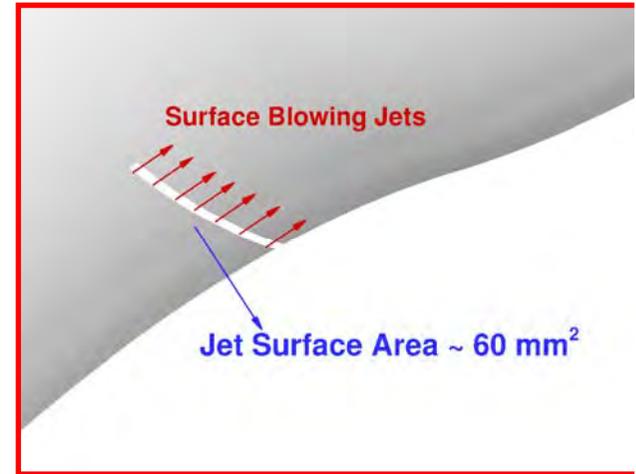
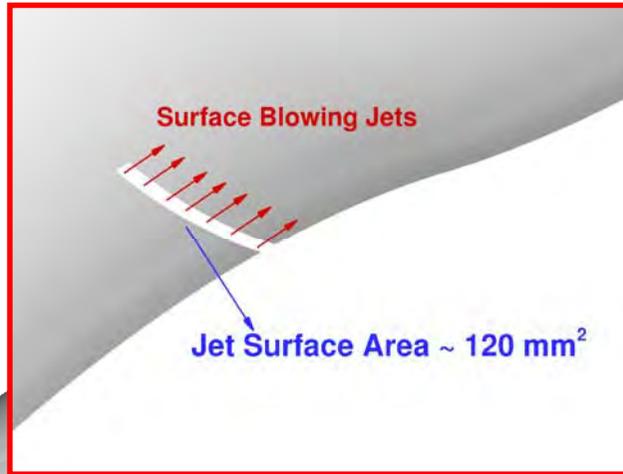
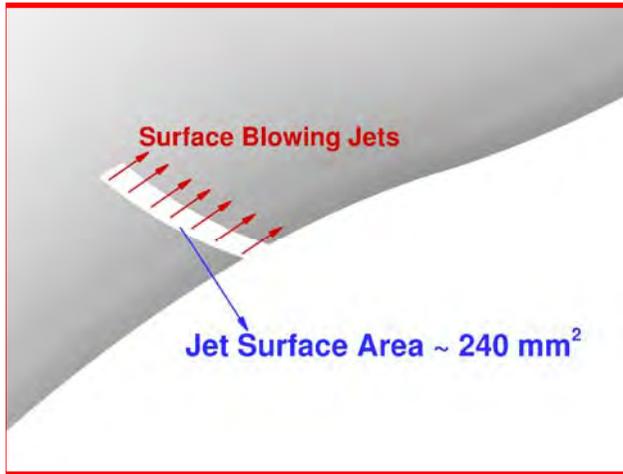


AIP Virtual total pressure rake at $M3A=0.4$
Overset Grid
The Boeing Company/CSULB - RANS - SST Turbulence



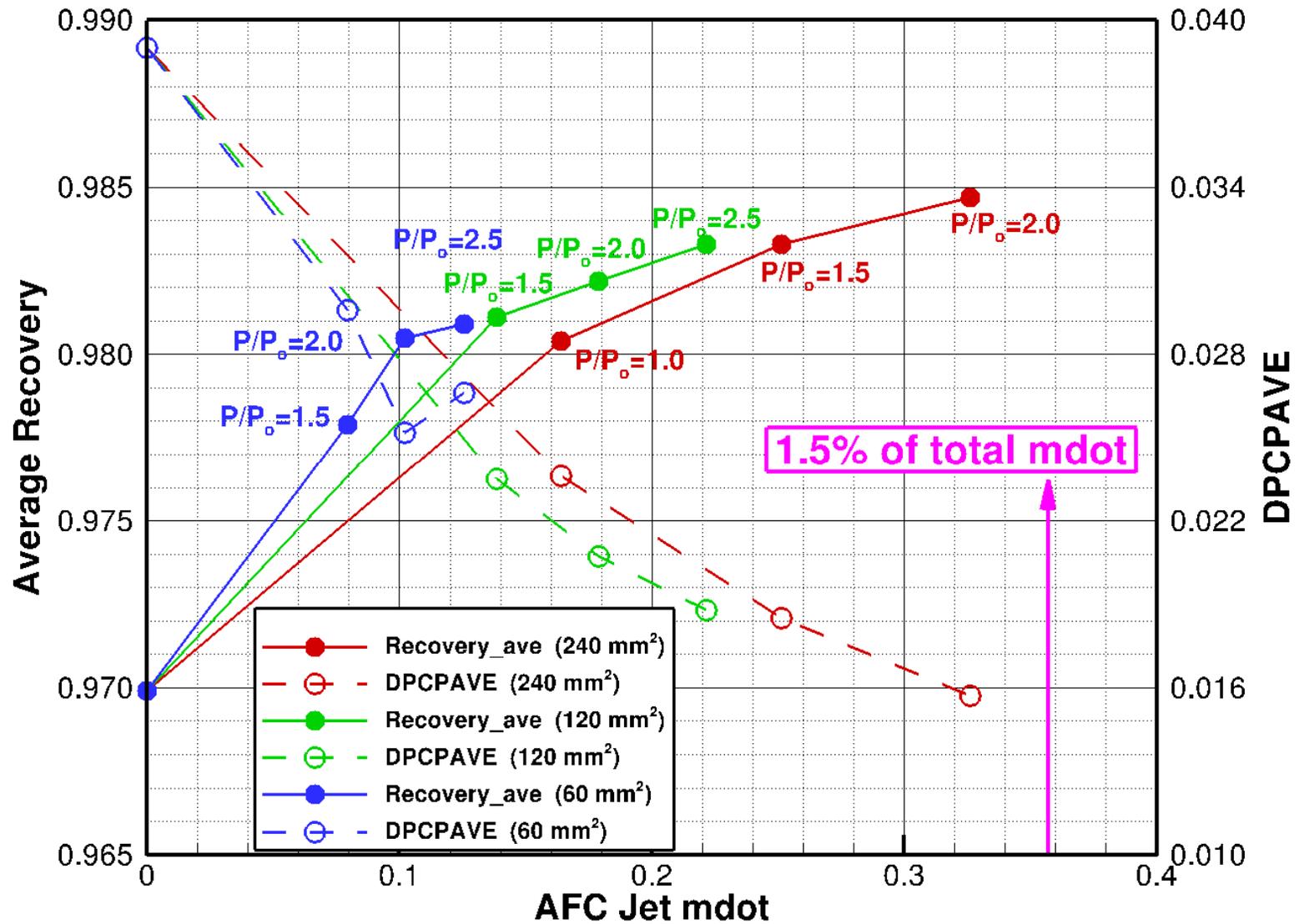
Active Flow Control – Surface Blowing Jets

Engineering, Operations & Technology



Active Flow Control – Surface Blowing Jets

Engineering, Operations & Technology

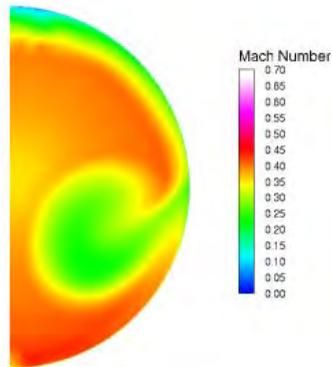


Passive vs Active Flow Control

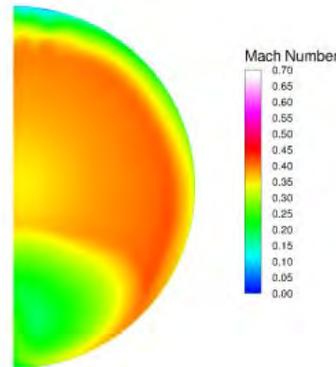
Passive vs Active Flow Control

Engineering, Operations & Technology

3 VGs
Recovery = 0.9777
DPCP_ave=0.0216



60 mm², P/P_o=1.5
Recovery = 0.9779
DPCP_ave=0.0296



120 mm², P/P_o=2.0,
Recovery = 0.9822
DPCP_ave=0.0207

